



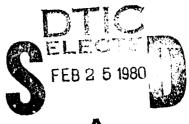
AN EVALUATION OF HUMAN THERMAL MODELS FOR THE STUDY OF IMMERSION HYPOTHERMIA PROTECTION EQUIPMENT



UCTOBER 1979

FINAL REPORT

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PREFACE

This report documents work conducted under Task Number 3 of Contract Number DOT-CG-72074-A from January 17, 1978 to August 21, 1979. The work was performed at Clemson University under the auspices of the U. S. Coast Guard, with LTjg Steven F. Wiker and Ens. John A. Budde serving as program technical monitors. The contract principal investigator was Dr. R. Michael Harnett. The leader of the part of the work addressed in this report was Dr. E. R. Baker, IV. The graduate assistant in this work was Jeffrey L. Ringuest.

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AN EVALUATION OF HUMAN THERMAL MODELS FOR THE STUDY OF IMMERSION HYPOTHERMIA PROTECTION EQUIPMENT

for

UNITED STATES COAST GUARD U.S. Coast Guard Headquarters

Contract No. DOT-CG-72074-A

Task Number 3

Draft Final Report, Part III

from

Clemson University

by

E. R. Baker, IV, Ph.D., R. M. Harnett, Ph.D., J. L. Ringuest

August 21, 1979

1.0 INTRODUCTION

The use of human subjects for testing hypothermia protection devices is not without risk. Additionally, such in vivo tests can be used only to mild state of hypothermia (rectal temperatures not less than 35°C). A mathematical model capable of accurately simulating the thermal responses of a protected man in a cold environment would be an attractive alternative to human experimentation. In addition, the time required to perform the human experiments makes this alternative method of device evaluation extremely attractive.

1.1 Background

A variety of models have been proposed for the human thermal system.

Some of the models have been established from experimentation while others have been constructed from the theories of thermodynamics and fluid mechanics.

Some models predict the behavior of the whole body while others are specialized

to a particular body segment. The interest in this research lies in models of the entire human body. Excellent reviews of the work in this field are available by Hardy (1972), Mitchell, et al. (1972), Shitzer (1972), Fan, et al. (1971), and Hwang and Konz (1977).

Mathematical models of the whole human body may be generally classified as single cylinder or multi-segment models. Further classification is accomplished by determining whether thermoregulation is internal or external to the model. Models with internal thermoregulation functions may be viewed as a composite of two submodels, one for the passive system (physical system) and one for the controlling system.

The complexity of models of the passive systems depends upon the number of body segments modeled, their geometries, the number of nodes and shells (layers of segment composition) attributed to each segment and the sophistication of the model circulatory system. The complexity of controller models range from simple function-evaluation types to those which compute error signals based on variables such as average skin temperature, core temperature and skin heat flux. All controllers determine metabolic sate and in some models, the control system also determines the sudometer (sweating) and vasometer (variable blood flow) responses.

The models are most generally expressed as a set of differential equations. Early models were solved using analog computers. The advent of larger, faster digital machines has resulted in most models now being programmed for digital computers. The solution methodology most often employed is that of finite differences.

1.2 Objectives

The objective of this research was to determine the feasibility of using an existing human thermal model in the evaluation of immersion hypothermia protection devices. To accomplish this task a review of the literature was conducted leading to the selection of candidate models representing the general types available. Computer codes for these models were obtained and modified as necessary to implement them on the computer system at Clemson University. These modifications of the computer-based models were not intended to accomplish basic structural changes to the

models or to improve their intrinsic capabilities.

The selected candidate models were to be validated against the data collected from the human immersions discussed in Part I of this report, see Harnett, et al. (1979). Based on the results of these validations, recommendations were formulated ragarding the potential usefulness of each selected model. Also, the most important areas for model improvement were determined and recommendations regarding their priorities were developed for consideration prior to undertaking any efforts aimed at improving the capabilities of a model.

1.3 Scope

The scope of this effort included only an evaluation of existing models. No new model development or modification of existing models was required. However, in the course of implementing the computer-based models and experimenting with them, it was also possible within project cost and schedule constraints to develop and implement a number of substantive model modifications. These modifications improved the performance of the affected models and in some cases were necessary in order to give the models any chance of performing as required for the evaluation of cold water protection equipment.

2.0 MODEL SELECTION

After a review of the literature it was decided that this investigation would be limited to non-steady state models containing controllers. This decision was predicated on two facts: (I) the human response to cold water immersion was unlikely to be steady state and (2) we would be unable to externally control the model since no experimental immersion data was available which correlated metabolic rate, vaso-constriction and surface blood flow to the physical parameters of the model such as core, skin and water temperatures.

References were found in the literature to but a few models fitting these criteria. Conversations and correspondence with the authors of these models revealed several other unpublished models. However, these models were all reported to be very similar to those found through the literature search. The following five models were obtained: Stolwijk, (1972); Montgomery, (1972); Gordon, (1972); Kuznet, (1974); and Winton and Linebarger, (1971). An additional model, without a controller, Wissler (1966) was requested from its author; but this request was not granted.

These models represented several different philosophies in the modeling of both the passive and control systems. Stolwijk's and Kuznet's models had been used primarily for investigation of hyperthermia. The applications were in support of the NASA manned space flight program. Both models were implemented on Clemson's IBM 370/165 and exercised with test data as specified by their authors. Neither model, however, was modified for further evaluation. Stolwijk's model was dropped from further investigation because Montgomery's model was determined to be an extension of it which had been used for diving studies. Kuznet's model was dropped because of its similarity to both the Gordon and Montgomery models. The three remaining models, Winton's, Montgomery's and Gordon's were chosen because of the contrasts among them in terms of their general approaches and complexity.

The Winton and Linebarger Model

The Winton and Linebarger (1971) model is the simplest of the models studied. This model was intended for study of both the steady-state and transient response of the human thermoregulation system to various degrees of internal and external thermal stress. Emphasis in this model was placed

on the feedback structure and controller mechanisms involved in thermoregulation. This model has been exercised using both analog and digital simulation.

Winton and Linebarger represent the shape of the body as a cylinder having three concentric layers. The inner-most layer represents the core of the body, which is composed of the deep tissues and internal organs. Surrounding the inner core is a middle layer made up of muscle and fat. The skin comprises the outer layer of the cylinder. For purposes of analysis the thermal properties of the three-layer model are represented by an analogous electrical circuit. The thermal system and its electrical analog are governed by identical differential equations which form the basis of this model.

Three primary control mechanisms are included in this model: sudometer, vasometer and metabolic. These control mechanisms are incorporated into the model by varying the related parameters of the circuit analog in an appropriate manner. Physiological studies have shown the importance of both core and skin themperatures in thermoregulation. Based on these studies feedback signals from the core and skin have been included in this model.

The Montgomery Model

The Montgomery model is an extension of the Stolwijk (1970) model intended to allow investigation of heat loss/gain during underwater diving work. Its thermoregulatory system is divided into two distinct subsystems: the physical-controlled subsystem and the dynamic-controlling subsystem. The controlled subsystem consists of the physical portions of the body.

The controlling subsystem contains the central hypothalamic thermointegrator, the central set point temperature and associated afferent and efferent signal pathways. The controlling subsystem receives afferent signals from all partions of the body, integrates the signals, compares the results to the central set point and distributes the appropriate effector command signals to all portions of the body.

The controlled subsystem, Figure III-1, consists of the head which is considered a sphere and cylinders representing the trunk, arms, hands, legs and feet. Both arms, hands, legs and feet are represented, because of symmetry, by one

FIGURE 111-1 SCHEMATIC OF THE MONTGOMERY CONTROLLED SYSTEM

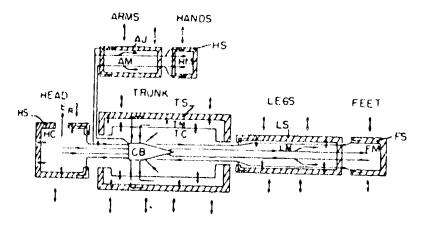
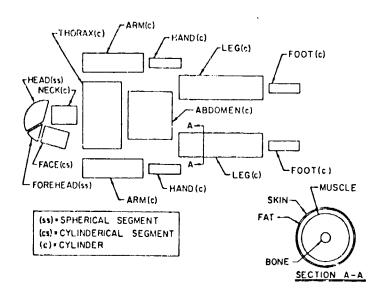


FIGURE 111-2 SCHEMATIC OF THE GORDON CONTROLLED SYSTEM



Body elements that comprise the possive system

cylinder each. Each body segment is composed of eleven concentric layers: four representing the body core, four representing the muscle layer, and one each representing the body layers of fat and skin, and one representing the outer wet suit layer. A central blood compartment simulates the body's central blood pool which exchanges heat with all other body compartments, via convective heat transfer, through simulated blood flow to each compartment. Each of sixty-one body compartments is represented by a heat balance equation which accounts for internal heat generation, conductive heat transfer between adjacent compartments, and convective heat exchange with the central blood compartment. Where applicable, respiratory heat generation and heat loss are included in the body compartment heat equations. Additional heat balance equations represent a wet suit on the six body segments and include the effects of conductive heat transfer with the skin and conduction-convection with the ambient water. Each of the sixty-seven heat balance equations includes the thermal capacitance of the compartment enabling the transient response of the compartment to be simulated.

The Gordon Model

The Gordon model is the most complex of the three models examined. Gordon also divides the temperature regulatory system into two major subsystems: the passive subsystem and the control subsystem. The passive subsystem is divided into ten body segments. Cylinders are used to represent the neck, thorax, abdomen, arms, hands, legs and feet. A cylindrical segment represents the face and spherical segments represent the head and forehead. All segments consist of four concentric layers. Both the number of segments and the number of integration nodes per layer is variable. The model was implemented at Clemson with the segmentation described above and eleven nodes per segment. Both of these choices were used by Gordon in his initial validation of the model. They represent a logical balance between computational burden and numerical precision.

Since the body is modeled as concentric spherical or cylindrical shells, (Figure III-2) the governing equations were obtained by considering a shell of uniform properties. An energy balance equation is used for each shell. To complete the passive subsystem model, heat transfer involving the blood pool was modeled, including counter-current heat exchange between certain body elements

and the blood pool. The partial differential equations (which from the energy balancing) are nonlinear and are solved by finite difference techniques.

The control subsystem in this model generates several "error signals" which are deviations of average skin heat flux, average skin temperature, and hypothalmus temperature from set-point values obtained passively during basal conditions (resting with no food in stomach). Combinations of these error signals are weighted to produce the control subsystem signal. The equations which describe this control signal make up the control system model. Although the original emphasis of this model was to simulate exposure to cold air, provisions were made for the future addition of a warm-environment controller. The computer code for the Gordon model was written in FORTRAN.

3.0 THERMAL TESTING PROCEDURE AND RESULTS

Any model to be used for evaluations of a cold-protection device must be provided a description of the thermal properties of the device. The models considered in this study were modified to accept, as input data, the thermal conductivity (resistivity) of the devices. Since this data was not available for many of the devices included in this study, it was necessary to determine them through experimentation. This chapter presents a brief description of the testing methodology and the results obtained with it. A detailed description of the methodology was given by Baker (1979).

Briefly, the procedure was to monitor the temperature in a bag of water (initially warm) which had been surrounded by the test material, while it was vigorously agitated in a large tank of colder water. From this data, the BTU loss could be calculated. Assuming linearity over a briaf span (3-5 minutes) an estimate of the thermal resistance (R) of the material was calculated. Values obtained in this way are only approximations but were seen to compare well with published values obtained using the methodology of ASTM-C518-70.

Each device included in the cold-immersion test (except the PFD) was tested to determine its thermal resistance. A number of tests were performed on each device. Any test whose result was suspect due to obvious nonlinearity was discarded. This resulted in sample sizes ranging from three to five for individual protective devices. Table III-I presents thermal resistance values (R values) determined in this testing and coefficient of heat transfer values (U values).

The suits which relied upon Aramid underwear to provide thermal protection were tested with a swatch of it placed between the bag and the device being tested. These suits are identified in Table III-I with the comment (W/A). It was noted that in the case of the CWU-21/AP, the Arimid swatch was wet at the completion of each test. As was pointed out in Part I of this report, Harnett, et al. (1979), this suit was also observed to leak during its in vivo cold immersion testing.

The ILC industries prototype and Dr. Rentsch's prototype rely mainly on an inflatable air bladder to provide thermal protection. The conductivity

TABLE 111-1 RESULTS OF THERMAL TESTING

	THERMAL RESISTIVITY ("C.hr.m"/k cal)	IIVITY sal)	HEAT TRANSFER COEFFICIENT,
Test Article	Mean	Std. Dev.	(k cal/°C/hr/m²)
Bayley Exposure Suit (PVC foam)	0.1883	0.00654	5.309
Bayley WeatherMate Plus	0.09668	0.0210	10.343
Helly-Hansen Survival Sult (D-600-0)	0.1024	0.00876	9.765
Henderson Zip-On Exposure Sult (2080-4)	0.1222	0.03087	80
Henderson Prototype Jacket	0.1222	0.03087	8.18
*iLC industries Prototype Survival Suit (W/A)	0.07189-0.51005	0.00664	13.910-1.96
Medalist Ski Shorty (7010)	0.0686	0.0026	14.575
Mustang U-VIC Thermofloat (1661)	0.1803	0.0665	5.545
NADC Goretex Experimental Coverall (W/A)	0.1023	0.00607	9.775
*Dr. S. B. Rentsch's Prototype Survival Suit (W/A)	0.02949-0.7572	0.00759	33.909-1.321
S.1.D.E.P. Seastep Survival Suit	0.1269	0.008	7.875
Stearns Windjammer Jacket (FJ-55)	0.2189	0.0708	4.567
Stearns Offshore Survival Jacket (FS-500)	0.1669	0.0368	5.989
Stearns Heavy-Duty Offshore Survival Suit (FS-71)	0.1058	0.00896	9.452
U.S. Air Force Modified Anti-Exposure Assembly (CWU-21/AP) (W/A) 0.0455	0.0455	0.00607	21.59
Blue Denims (wet)	0.00673	0.000533	148.588

*First number is the vaiue determined experimentally for the deflated suit. The second number is the calculated value for the suit inflated. No standard deviation is shown for the calculated value.

of these two devices could not be determined by direct tests as previously described. Instead, we measured the intrinsic conductivity of the suits deflated. These values along with the average thickness of the air space with the suit inflated were used to calculate an overall conductivity. The calculated values shown in Table III-I are actually the average of two calculated values. One value was calculated assuming that the air space was devoid of convection; the second assumed that the convection-conduction was equivalent to that between two smooth parallel plates. The first assumption leads to an optimistic value, the second to a pessimistic value. The average was taken with uniform weighting. This approximation is necessitated by the geometry of the problem which does not lend itself to solution by known techniques.

The accuracy of this method is unknown. It is suggested that in the future, conductivity tests be run using a heat balance system similar to that recommended in ASTM-C518-70 but adapted to testing in "the wet". For example, a brass sphere containing water and a heating coil could be covered with the material to be tested. By monitoring the temperature in the vessel during immersion, the power through the coil can be adjusted until a steady-state temperature is achieved within the vessel. The power being supplied to the coil is then directly related to the conductivity of the protective material.

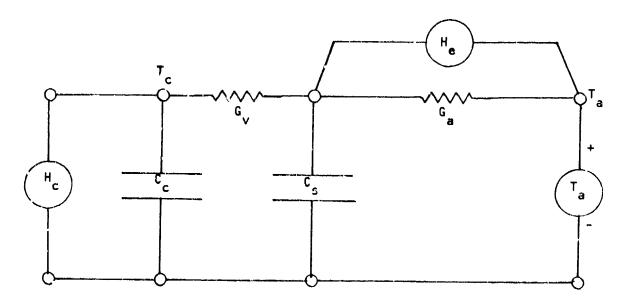
4.0 EVALUATION OF THE SELECTED MODELS

The models selected were modified to accept the inputs from the thermal testing described in the last chapter. This simplified the process of exercising the models for the variety of test articles considered in the study. It was also necessary to modify by adding a layer to represent the protection equipment. Further, It was necessary to modify the modeling of convective-conductive heat transfer at the interface with the environment to reflect the differences resulting from water as opposed to air immersion.

4.1 The Winton and Linebarger Mode!

The Winton and Linebarger model is the simplest of the three selected for evaluation. The three-layered cylinder used to model the body was represented by the electrical circuit analog shown in Figure III-3.

FIGURE 111-3
WINTON-LINEBARGER 3-LAYER MODEL



The model, originally implemented on an analog device was later written in CSMP. A CSMP version of the model was implemented and evaluated in this study.

In Figure III-3, H_{c} is a current source which represents basal metabolic

thermogenesis, shivering thermogenesis and respiratory heat loss. $G_{\rm v}$ is a resistor representing the net heat conduction of the body tissues and $G_{\rm a}$ represents the net conduction from body to environment. $C_{\rm c}$ is a capacitor representing the heat capacitance of the body tissues except for the skin which is represented by $C_{\rm s}$. $H_{\rm c}$ is a current source representing evaporative heat loss. $T_{\rm a}$ is a voltage source representing ambient environmental temperature and $T_{\rm c}$ and $T_{\rm c}$ represent core and skin temperatures, respectively.

To accommodate modeling of a thermal protective device it was necessary to modify the circuit in Figure III-3. Figure III-4 shows the modified model.

T_c
T_s
T_s
T_s
T_w
G_w
G_a
T_a

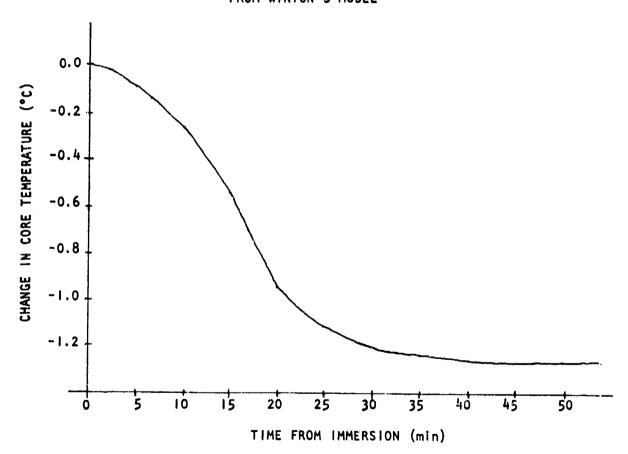
FIGURE 111-4
MODIFIED WINTON-LINEBARGER MODEL

Here G_W represents the conductance of the protective device, and C_W its thermal capacitance. The current source, H_W , was included to allow the modeling of devices which, themselves, produce heat. T_W is the temperature of the protective device. For immersion simulation H_W is set for zero. All of the protective devices considered in this study are passive, so H_W was set to zero. A listing of the CSMP program used to implement the model in Figure III-4 is shown in Appendix A.

The model performed well considering its simplicity. Core temperature

losses were of the same magnitude as those observed during a human immersion. The time trace itself however was not the same as is seen in Figure III-5. The model loses a significant amount of heat before the metabolic process can begin to compensate. The metabolic rate is driven as high as necessary to overcome the rate of heat loss and the model finally exhibits a plateauing.

FIGURE 111-5
TIME-TEMPERATURE PROFILE
FROM WINTON'S MODEL



The initial large loss of heat, as reflected by a rapidly declining core temperatura, is attributed in the main to the controllers inability to react to the high gradient experienced upon immersion. As with many models, Winton's was developed for use in estimating steady-state conditions. As a consequence, large continuing changes are required in core and skin temperatures before the metabolic rate becomes sufficiently elevated to

overcome the rate of heat loss due to the immersion.

Making the model more useful for evaluating protective devices would require modification of the controlling subsystem. Because the mode' represents the surface of the subject as a single cylinder, it is applicable only to the study of full body protective devices. Some generalizations to this model are possible. With modification of the controller good estimates of cooling rate may be possible for full body suits. Since the model requires small amounts of both computer time and storage, there may be sufficient justification to warrant this modification effort.

4.2 The Gordon Model

As described in Chapter 2, Gordon represented the body by a set of spherical and cylindrical segments consisting of concentric layers. Each layer was subdivided to finally represent each segment as a set of partial differential equations across eleven nodes. Each node was centered in a shell of uniform material. The resulting equations are solved using finite difference techniques.

The addition of a protective device to this model required only two small changes to the model. An additional layer was modeled for each body segment covered by the device. Each segment originally had four layers. Because of the rapid transient temperatures to be experienced in this layer, it should be modeled with not less than two integration nodes, Wissler (1971). Thus, at a minimum, the model would consist of thirteen differential equations for each protected segment.

The second change required was in the control subsystem. In the original model, the controller is hooked to the last integration node, that of the skin. Since the model was built to accept a varying number of nodes per segment, the addition of the protective device would cause the controller to use as input temperatures and fluxes calculated for the furthest node from the segment core which would be protective device surface temperatures and fluxes. This is easily remedied in either of two manners. First, one may fix the number of nodes representing body in each segment thus affixing to the skin a constant node number. Alternatively the node number representing the skin may be calculated by including as input the number of nodes used to represent the protective device on each segment.

As mentioned above, the three models to be modified were first examined under conditions for which they had been designed. In this testing Gordon's model was found to be the best overall analog. Before modifications were made to model the protective device, the model was set up to simulate a nude cold immersion. As a result of these runs, a major problem was observed in the Gordon model. One would expect that, after an initial rise in core temperature, a decrease would occur which may or may not find an equilibrium level. The period of initial rise and the magnitude of the rise can vary. However, from experimental evidence one would expect the period to be relatively short with a magnitude of at most a few tenths of a degree celsius for a nude man in cold water. The Gordon model exhibited the initial increase upon immersion. However, the core temperature continued to climb as the simulated immersion continued. At the end of a 5-hour simulated immersion, the core temperature had reached approximately 42°C.

A re-examination of the model under the conditions for which it was validated by its author showed that the problem could have existed at that time. It was validated during relatively short (2 hour) simulations of exposure to cold air (5°C). Increasing core temperatures were expected during this period. Their appearance in the simulation results seemed to indicate that the model was working well. Unfortunately, the validations with cold air immersions did not extend into the time period when core cooling occurs. This problem was not observed in the initial testing at Clemson because the simulations were also of relatively short duration.

The Gordon model employs the most detailed modeling of the human physical structure of the three models examined. It was anticipated that it would prove to be the most accurate of the three selected models. For this reason, while it was not an objective of this effort to identify and correct modeling errors, an attempt to do so was made. The equations for the physical system were verified by rederivation. The coding was checked and several simplifications were made. However, the problem could not be identified. It was finally decided to terminate efforts to correct the Gordon computer code, in favor of concentrating on the Montgomery model.

4.3 The Montgomery Model

As suggested earlier, the Montgomery model is basically an adaptation

of Stolwijk's model designed for use in evaluating the thermal aspects of diving activity. The model divides the body into ten compartments/segments. Each segment is subdivided into eleven concentric layers. Each compartment represents a lumped thermal capacitance with appropriate modes of heat production and heat transfer to other compartments. Each body layer generates metabolic heat at a basic rate and exchanges convective heat with the central blood pool. The eleven compartments of each segment exchange heat via conductive transfer with adjacent compartments as function of layer geometry and tissue thermal conductivity. Each wet suit segment exchanges heat with the environment as a function of wet suit properties and ambient water conditions.

The starting point in the development of the thermal network for a given subject is to estimate his percent body fat from his height and weight. His total surface area is also estimated from his height and weight. The surface area of each segment is then calculated from the total surface area.

The relative weights of the various segmental layers are calculated from the subject's total body weight and a percentage weight distribution. The various compartment weights, when multiplied by the corresponding specific heat value, yields the thermal capacitance value for each compartment. Since the core of each segment consists of both skeletal and visceral tissue which have different specific heat values, they must be treated separately and averaged.

The central blood compartment, representing the blood in the heart and the great vessels, is assumed to contain 2.5 liters of blood. The thermal capacitance of the central blood compartment is subtracted from the total thermal capacitance of the trunk core. The metabolic heat generation in each body compartment is calculated using the distribution given by Stolwijk and Hardy (1966).

The convective heat exchange that takes place between each body compartment and the central blood pool as a result of blood flow is calculated using the basal blood flow values for each compartment. The thermal conductances of each compartment are assumed to be uniform, concentrated at the compartment's center of mass and only dependent upon compartment temperature. Thermal conductance between layers is a function of the thermal

conductivity of the material between compartments, the distance between compartments, the area of the heat transfer surface located at the midplane between compartments and the temperature of the two compartments.

The wet-suit-to-ambient-water heat transfer coefficients are dependent upon the geometric shape of the body segment, the ambient temperature and pressure; the viscosity, thermal conductivity, heat capacity and density of the surrounding water; and the water velocity relative to the body segment. Wet suit compartment thermal capacitance values are calculated as a function of wet suit specific heat and wet suit density for each segment.

The Stolwijk (1970) blothermal model was used to form a basis for the controlling subsystem. The first change that was made to the Stolwijk model was to provide more compartments to represent the core and muscle tissue of each body segment. Finite difference methods of solution using lumped nodes will produce errors when new gradients develop in the relatively thick muscle and core layers. This type of error may be decreased by introducing additional compartments in the core and muscle portions of the controlled subsystem. This method was used by Wissler (1964) to Improve the simulated response to cold exposure.

The core and muscle portions of each segment were divided into four compartments, each having one-fourth of the core of muscle mass of the given segment. An additional compartment was also provided to represent the wet suit covering each body segment.

The effect of evaporative heat loss from the skin compartments is negligible under diving or totally immersed conditions. The evaporative heat loss from the trunk core is equal to that amount of heat that is carried away from the body during expiration of the respiratory gas. The quantity of heat loss from the respiratory tract for any gas mixture can be calculated from the physical properties of the gas mixture and the thermal and dynamic characteristics of the respiratory system. Respiratory heat loss is proportional to the respiratory minute volume, which is in turn related to the amount of oxygen required to provide energy for metabolic needs. The respiratory loss of heat is somewhat offset by the work of breathing. the net amount of evaporative heat loss from each of the trunk core compartments is taken to be one-fourth of the difference

between the total respiratory heat loss and the total heat generation due to respiration.

Net heat flow into or from each compartment is then calculated. The skin compartment in each segment loses, through conduction to the wet suit, an amount of heat equal to the heat transfer coefficient multiplied by the temperature difference between the skin and wet suit compartments. Since a diver does not receive solar heat input and does not transfer radiant heat to his surroundings, the environmental heat transfer coefficients used by Stolwijk (1970) have been replaced by convective-conductive heat transfer coefficients between the wet suit and ambient water. The water-neoprene skin surface heat transfer coefficient is dependent upon the geometrical shape of the body segment; the ambient temperature and pressure; the viscosity, thermal conductivity, heat capacity and density of the surrounding water; and the water velocity relative to the body segment.

Little modification was necessary to the physical system of Montgomery's model since it already had provisions for an additional layer between the skin and environment. Initial modifications were directed toward increasing the flexibility of device modeling to allow simulation of other than full body suits. In addition, modifications suggested by Montgomery (personal communications) were implemented to simulate immersion of the body with the head and neck (modeled as a single segment) exposed to air. These changes were made in the WETMAN subroutine of the model computer code.

Typical Results with Montgomery's Model

Runs were made to simulate several of the devices included in the cold immersion test described in Part I of this report, Harnett et al. (1979). The results of one series of these simulations is shown in Figure III-6. The article modeled (WP3) was a jacket-type device providing protection basically to the trunk and arms. The figure presents the simulation results, for a man 172 cm tall weighing 74.4 kg, and the experimental observations for five volunteers testing this device.

Considerable variation in response is exhibited by the five experimental observations, largely due to somatotype differences among the subjects. None of the individual responses is represented well by the model results. The model's metabolic control subsystem initially fails to recognize the heat drain caused by the cold immersion. This results in a too slow increase in the rate of thermogenesis and a rapid cooling shortly after immersion.

110 120 130 140 150 160 "£ Model 8 TIME FROM IMMERSION (MIN) FIGURE III-6 RESULTS WITH MONTGOMERY'S MODEL AND FIVE COLD !MMERSIONS -8 - 6 2 . 09 9 E - 8 ¥. 9 2 20 2 +9.0+ +0.4 ا . -----+0.2 4.0--0.6 -0.8 -1.2 -0.2 0 CHANGE IN RECTAL TEMPERATURE (°C)

₹/

When the metabolic controller does recognize the loss it elevates the metabolic rate by large steps to compensate for the decline in average $skin^{\frac{1}{\alpha}}$ and head core temperature. The result is a complete stabilization of core temperature.

Montgomery's Metabolic Rate Controller

The controller for metabolic rate formulated by Stolwijk and used by Montgomery calculates changes in metabolic rate based upon the following formulation.

$$\Delta MR = C_{c} \cdot (\Delta T_{c} + R_{c} \cdot T_{c}^{\prime}) + C_{s} \cdot (\Delta T_{s} + R_{s} \cdot T_{s}^{\prime}) + C_{c} \cdot (\Delta T_{c} + R_{c} \cdot T_{c}^{\prime}) \cdot (\Delta T_{s} + R_{s} \cdot T_{s}^{\prime})$$

where ΔMR = change in metabolic rate C_c , C_s , C_s , R_c and R_s are weighting factors

 ΔT_c = deviation of head core temperature from a set point value

 T_c' = rate of change of head core temperature

 ΔT m deviation of average skin temperature from a set point value

 T_s^* = rate of change of average skin temperature

The weighting factors (constants) are defined as follows.

$$R_c = 0$$

$$R_{s} = 0.03$$

$$C = 21.0$$

The definitions of C_c , C_s and C were used by Stolwijk based upon the experimental evidence of Benzinger, et al. (1963). With these definitions the metabolic rate controller simplifies to the following.

$$\Delta MR = 21 \cdot \Delta T_c \cdot (\Delta T_s + 0.03 T_s^{-})$$

^{*}Average skin temperature is defined as the average of the outer surface temperatures of each compartment, weighted by their proportional amounts of surface.

This simplified model is completely insensitive to the rate of head core cooling and is not particularly sensitive to the rate of skin cooling. The model is largely based on the amounts of cooling in these two temperatures and places essentially equal emphasis on the two. Initially, there will be a rapid decrease in average skin temperature but very little change in core temperature. This implies that the product in the above equation will be small, resulting in small increases in the metabolic rate, until there has been a significant decrease in head core temperature.

This helps explain the behavior of the model as depicted in Figure 111-6. It is obvious then that there is considerable room for improvement in modeling metabolic control. While not included in the scope of this project, the development of a new controller was seen to be essential to a positive finding that a model can serve the purpose addressed in this study.

The validity of Montgomery's (Stolwijk's) metabolic controller for certain conditions (e.g., a nude man in cold air) has been shown by many people including Stolwijk. It was, as stated above, based on experimental data. It was decided, therefore, to formulate a new controller for the immersion environment based on the data collected in the human immersion portion of this study summarized in Part I of this report.

Improved Metabolic Controllers

Relevant data available included rectal temperature at 15 cm, skin temperatures at the toe, thigh, forearm, bicep, groin and subscapular sites and periodic measurements of metabolic rate. The procedure used to develop the controller was to establish, through regression analysis, linear models relating changes in metabolic rate to changes in the rectal and skin temperatures. The basis of comparison for determining these changes were measurements made following a 30-minute rest period prior to commencing cold immersion.

The initial attempt was based upon regression analysis applied to the pooled sample of observations obtained in the laboratory. When the resulting metabolic controller was implemented in Montgomery's model, fair predictions of cooling rates resulted for most of the wet-mode suits but the predictions for the abandon-ship type dry suits were much worse.

The experimental data was then segregated in two subsets -- one obtained during immersion in 11.8°C water with wet-mode suits and one obtained during

immeration in 1.7°C water with dry-mode suits. These data sets were analyzed separately to produce a metabolic controller for each condition. For the wet-mode suits the following model was obtained.

$$\Delta MR = 7.004 - 5.822 \cdot \Delta T_{th} = 2.407 \cdot \Delta T_{f} = 37.382 \cdot \Delta T_{r}$$
 where $\Delta T_{th} = change in thigh temperature
$$\Delta T_{f} = change in forearm temperature$$

$$\Delta T_{r} = change in rectal temperature$$$

This model had a correlation coefficient (R- value) of 0.77. The accuracy with which regression relations conform to the data is often expressed by "F test statistics". The significance of these statistics may be interpreted by the "level of significance" at which the hypothesis (that the observations follow the model) may be rejected. Small levels of significance indicate that the regression conforms well to the observations. The "level of significance" for this regression was 0.001.

The control model obtained from regression analysis of the data obtained with the dry-mode suits is the following.

$$\Delta MR = 1.987 - 2.654 \cdot \Delta T_{t} - 4.595 \cdot \Delta T_{th} - 5.361 \cdot \Delta T_{b} - 8.920 \cdot \Delta T_{c}$$
where $\Delta T_{t} = \text{change in toe temperature}$

$$\Delta T_{b} = \text{change in bicep temperature}$$
other symbols as previously defined

This model had a correlation coefficient of .85 and a "level of significance" of .0001.

These two relationships were implemented in the Montgomery model. It was necessary to accept some approximations in marrying the list of variables required by the controllers with those available in Montgomery's model. The variables were matched as follows.

Controller Variable

Toe temperature Thigh temperature Bicep temperature Rectal temperature

Model Variable

Average foot temperature Average leg temperature Average arm temperature Trunk core temperature

These approximations are unavoidable because of the simplifications involved in modeling the physical structure of the body in Montgomery's model.

Model Validation

The question of model validity is dependent upon the use to which the model is to be put. The objective in this case is to use the model rather than human experimentation as the basis to estimate the survival time associated with new developments in protection equipment. If one accepts the survival time model and prediction procedure presented in Part 1, then all that is required of the model is a prediction of core cooling rate which may than be used to estimate survival time. This would relieve the need to be particularly concerned with absolute temperatures predicted by the model for various body sites or transient aspects of their profiles.

Based on this method for estimating survival tima, model validity may be determined by establishing the accuracy of its predictions of the rate of core cooling. This may be done by performing statistical tests comparing the rates observed with the volunteer test subjects (Part I of this report) to corresponding rates predicted for them by the model. This data is naturally "paired" and so lends itself to paired analysis as a means of variance reduction. The "paired t test" described by Steel and Forrie (1960) was used for this purpose. The procedure is illustrated below. The test was run at the 0.05 level of significance with a two-tailed rejection region.

Null Hypothesis (H_O): There is no difference between mean simulated and mean experimentally observed cooling rates

	Predicted Cooling Rate(°C/hr)		Deviation (d)	
Subject	Simulation	Experimental	Simulation - Experimental	
MK	1.037	2.510	-1.473	
GE	1.472	1.800	-0.328	
TP	1.100	0.556	0.544	
вн	.697	0.391	0.30	
PK	.621	0.244	0.376	
		l		

$$\Sigma d^2 = 2.808$$
 $\Sigma d = 0.575$ $n = 5$ (No. of pairs)
$$\bar{d} = d/n = 0.115$$

$$S_{\bar{d}}^2 = \frac{\Sigma d^2 - \frac{1}{n}(\Sigma d)^2}{n(n-1)} = 0.137$$

$$S_{\bar{d}}^2 = 0.367$$

$$t-test statistic = \bar{d}/S_{\bar{d}}^2 = 0.310$$

Critical value of t at 0.05 level of significance with 4 degrees of freedom = 2.776

Since the calculated statistic is (ess than the critical value we cannot reject ${\rm H}_{\rm O}$.

A summary of these tests for each of the devices included in the cold immersion testing, except for the PFD is presented in Table III-2. From the table we observe that the model performed well for all of the wet-mode suits.

Those suits for which we must reject the hypothesis of sameness between model and observed average cooling rates are largely the abandon-ship suits. Reference to Figure I-I in Part I of this report will help explain the models failure. The model "sees" all simulated immersions with the subject completely underwater from the neck down. As can be easily observed from the pictures of the flotation attitudes of these suits in Figure I-I, much of the upper surface areas covering the legs, arms and trunk of each of these suits is exposed to air. The heat transfer coefficient for air is much less than that of water. One would therefore expect the cooling rates predicted by the model for these devices to be greater than that observed in the

TABLE III-2
RESULTS OF PAIRED t TEST
WITH REGRESSION-BASED CONTROLLERS

TEST ARTICLES	Calculated Statistic	Critical Value @ 0.05 Level	Dagrees of Freedom	Accept or Reject
Bayley Exposure Suit (PVC foam)	5.367	2.776	4	Reject
Bayley WeatherMate Plus	1.599	2.776	4	Accept
Helly-Kansen Survival Sult (D-600-0)	4.789	2.571	5	Reject
Henderson Zip-On Exposure Suit (2080-4)	5.237	2.776	4	Reject
Henderson Prototype Jacket	1.130	2.776	4	Reject
ILC Industries Prototype Survival Suit	0.387	2.776	4	Accept
Medalist Ski Shorty (7010)	8.172	2.776	4	Reject
Mustang U-ViC Thermofloat (1661)	0.009	2.776	4	Accept
NADC Goretex Experimental Coverall	9.102	2.571	5	Reject
Dr. S. B. Rentsch's Prototype Survival Suit (without respiratory heat reclamation)	1.289	2.776	4	Accept
S.I.D.E.P. Seastep Survival Suit	15.931	2.776	4	Reject
Stearns Windjammer Jacket (FJ-55)	0.416	2.776	4	Accept
Stearns Offshore Survival Jacket (FS-500)	1.823	2.571	5	Accept
Stearns Heavy-Duty Offshore Surv!val Sult (FS-71)	6.068	2.776	4	Reject
U.S. Air Force Modified Anti-Exposure Assembly (CWU-21/AP)	2.187	2.776	4	Accept

experimentation, as is the case.

The Stearns Heavy-Duty Offshore Survival Suit and the Henderson Zip-On Exposure Suit failed to pass the test. Reference to Figure I-I shows the same type of flotation attitude for these suits as for the abandon-ship suits. In addition to the above-stated reason, the inadequacy of the controller may be contributory to their failure.

Model results for two additional suits failed the paired t test. In both cases the model-predicted cooling rate was faster than that observed from our testing. Again reference to Figure i-I shows that both of these devices were tested with the subjects wearing a "water wings" type flotation device which exposed a significant amount of the trunk to the air. Additionally, the flotation device allowed the subjects to keep their arms and hands out of the water. Thus a faster predicted cooling rate from the model seems very reasonable.

Three suits passed the test which, by the above arguments, should have failed, the ILC prototype, Dr. Rentsch's prototype and the U. S. Air Force's CWU-21/AP. All three of these had simulation predicted cooling rates slower than we expected. For the ILC and Dr. Rentsch's prototypes this is most probably due to error in the estimation of their thermal conductivities. The estimated conductivities are, therefore, probably smaller than the reality. Thus while the model saw a completely submerged suit, it also saw a thermal resistance probably larger than reality. The combination of these two "errors" acted to cancel each other.

The average of the observed cooling rates was higher than the average of the model predictions for the CWU-21/AP. This suit, while ostensibly a dry suit, was observed to leak during testing, as noted in Part I and in Chapter 3 of this part of this report. Since the only thermal protection was the thin dry shell and arimid underwear, it is reasonable to assume that the major portion of the thermal protection offered by the underwear was lost when it became wet. Thus the model may have expected more thermal resistance than probably existed in the experimentation.

If one is concerned only with the prediction of cooling rates, the model, when used on suits that do not expose a great deal of body/device surface to air, appears acceptable. Overall, cooling rates predicted by

the model are faster than observed. Thus, in general the cooling rate obtained by simulation will lead to conservative estimates of survival time when used with the survival-time prediction model presented in Part I of this report.

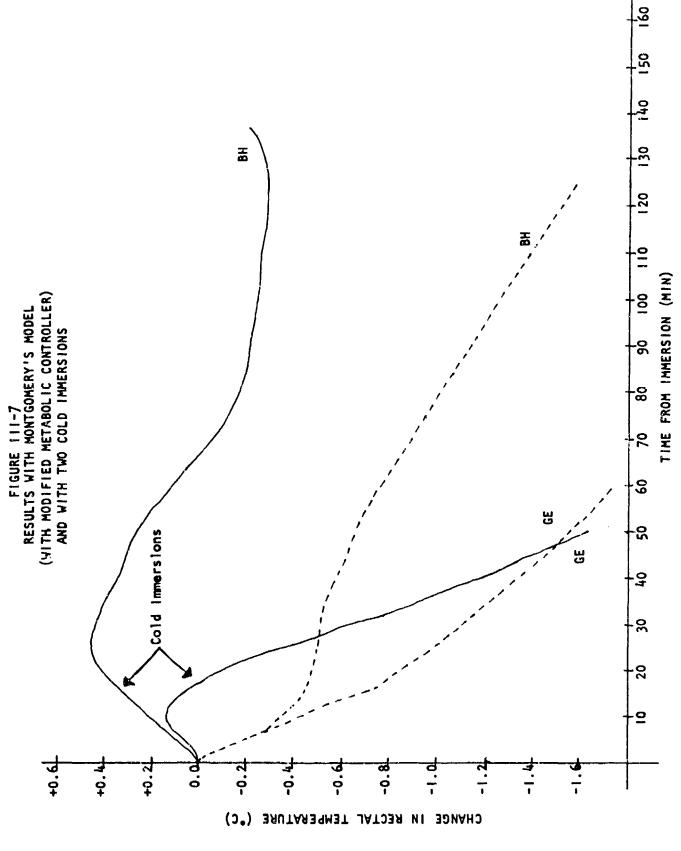
The failure of the model lies in its inability to accurately predict time traces of body temperatures. Figure !!!-7 shows a plot of rectal temperature as a function of time for two subjects wearing the U-VIC Thermofloat. The solid lines are plots from experimental data. Broken lines show the results obtained when the physical parameters of the two subjects were put into the model. Two important observations may be made. First, there is a complete absence in the simulation results of any initial rise in rectal temperature. Second, the average slopes of the observed and simulated rectal temperature traces (from maximal temperature to last observation) are very nearly equal. In fact, if the model traces are displaced to the right to coincide with the return of the experimental trace to entering temperature, the simulated and observed traces correspond very well. Since the slopes are approximately the same, the predicted cooling rates will be very similar. Thus the model was able to pass the paired t test even though the predicted and observed rectal temperature behaviors varied notably.

Tuning the Metabolic Controllers

As noted in Part I, for two of the human immersion tests we had the use of a Waters continuous reading oxygen consumption meter computer (MRM-I). The time profiles of metabolic rate observed using this device are seen in Figures I-4 and I-5 in Part I. It may be seen that immediately upon immersion the metabolic rate jumps by about 40 kcal/hr above a resting rate of approximately 70 kcal/hr. Since even the revised metabolic control models did not show this immediate increase, an experimental controller was constructed which included it as a constant. The resulting experimental controller for the wet-mode devices is the following.

$$\Delta MR = 40 + 1.237 - 5.339 \cdot \Delta T_{th} - 4.547 \cdot \Delta T_{f}$$

The last three terms of this experssion were obtained by regression and had an R-value of .69 and a level of significance of .04. For dry-mode suits tested in 1.7°C water, the 40 kcal/hr constant was added to the regression



equations determined above for these suits.

Figure III-8, shows the profile of change in rectal temperature for two subjects wearing the U-VIC Thermofloat. The observations are shown as solid lines while the simulation results are represented by the broken lines. The agreement between observation and simulation is much improved over that depicted in Figure III-7 for the modified controller based solely on regression.

Table III-3 presents the results of paired t test carried out between the simulation predicted cooling rates using these experimental controllers and those observed experimentally. The results are much the same as those presented for the modified controllers in Table III-2. The notable changes are the ILC prototype and Rentsch's prototype which fail now, exhibiting slower predicted than observed cooling rates as would be expected with a higher average metabolic rate. The increase in metabolic rate also helped the S.I.D.E.P. Seastep and Stearns Heavy-Duty Offshore Survival suits to pass the test. Overall, while these controllers are not as firmly supported by experimentation, the results are subjectively more satisfying.

Figure III-8 presents the profiles of three skin temperatures predicted by the model for an "average" man, as described in Part I of this report, versus the average results obtained experimentally. The experimental points are bracketed by one standard deviation. The profiles show only the first fifty minutes of the immersions. This is necessitated by the removal of one of the subjects at that time.

The profile of leg temperature predicted by the model, as shown in Figure III-8(a), lies outside of the bounds of the experimental data. However, the temperature plotted for the model is an average leg temperature while that plotted from experimental data is a thigh temperature. One would certainly expect an average leg temperature to be lower than a thigh temperature.

The profile of arm temperature is shown in Figure III-8(b). The initial deviation is due most probably to the rapid decrease in skin temperatures when the protective device initially floods with water. The model does not see this flooding and the simulated temperatures drop more slowly. In the model heat must be lost through the simulated device to the water. The frequent movement of the subjects (e.g., shivering) during the immersion helps to maintain some continual flushing. Thus the average arm temperature from the model would be expected to be somewhat warmer than that observed.

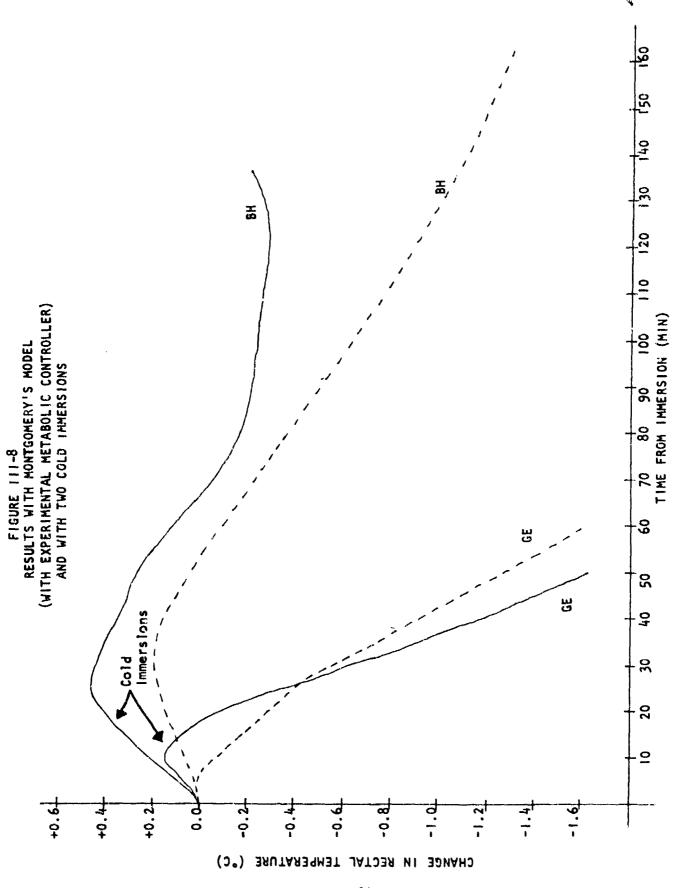
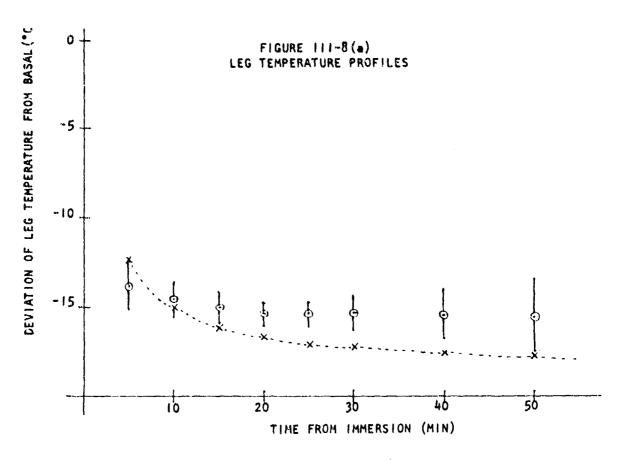


TABLE 111-3
RESULTS OF PAIRED t TEST
WITH EXPERIMENTAL REGRESSION-BASED CONTROLLERS

TEST ARTICLES	Calculated Statistic	Critical Value @ 0.05 Level	Degrees of Freedom	Accept or Reject
Baylay Exposure Sult (PVC foam)	4.088	2.776	4	Reject
Bayley WeatherMate Plus	2.03	2.776	4	Accept
Helly-Hansen Survival Suit (D-600-0)	2.926	2.57!	5	Reject
Henderson Zip-On Exposure Suit (2080-4)	4.552	2.776	4	Reject
Henderson Prototype Jacket	0.945	2.776	4	Accept
ILC Industries Prototype Survival Suit	4.750	2.776	4	Reject
Medalist Ski Shorty (7010) with Flight Suit	3.143	2.776	4	Reject
Mustang U-VIC Thermofloat (1661)	0.310	2.776	4	Accept
NADC Goretex Experimental Coverall	4.369	2.571	5	Reject
Dr. S. B. Rentsch's Prototype Survival Suit	14.609	2.776	4	Reject
(without respiratory heat reclamation) S.I.D.E.P. Seastep Survival Suit	2.597	2.776	4	Accept
Stearns Windjammer Jacket (FJ-55)	1.170	2.776	4	Accept
Stearns Offshore Survival Jacket (FS-500)	1.178	2.571	5	Accept
Stearns Heavy-Duty Offshore Survival Sult (FS-71)	1.706	2.776	4	Accept
U.S. Air Force Modified Anti-Exposure Assembly (CWU-21/AP)	1.323	2.776	4	Accept



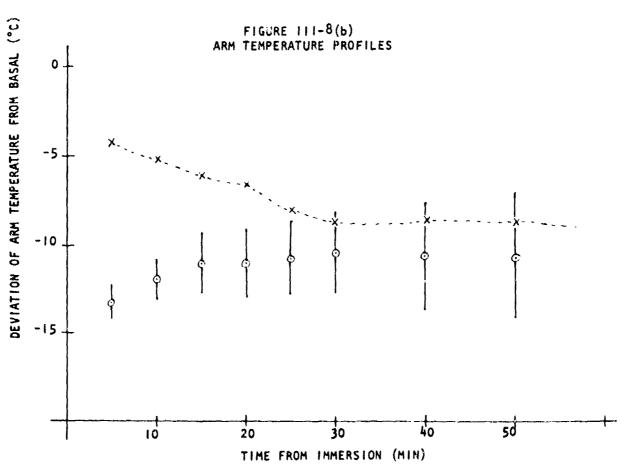


FIGURE 111-8(c)
TRUNK TEMPERATURE PROFILES

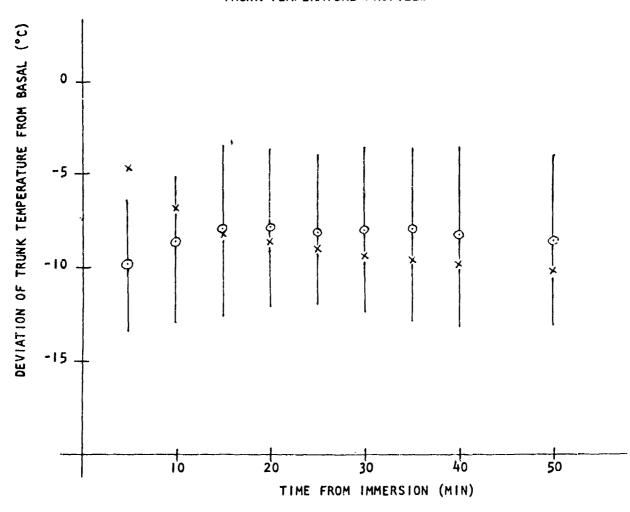


Figure III-8(c) shows the relationship between model predicted and observed temperatures for the trunk. Experimentally the trunk temperature was taken to be that monitored at the subscapular site. Again an unitial deviation may be seen which is easily attributed to initial flooding of the suit. In the long run the experimental profile is at a slightly higher temperature level than the simulated one. The simulated data is for the average trunk temperature which might reasonably be expected to be cooler than the subscapular site.

5.0 CONCLUSIONS AND RECOMMENDATIONS

The potential usefulness of a simulation model for evaluation of an immersion protective device is without question. In addition to eliminating the risk associated with human experimentation, there are economic edvantages in conducting evaluations by simulation rather than by human experimentation. Furthermore, simulation of an immersion may be accomplished in a short time. The computer resources required by the model are small, most will run in under one minute of CPU time on an IBM 3035 and use less than 256k of core. The cost of a typical run made with Montgomery's model in this study was approximately \$7.80. Obviously this is a much more economical approach than direct human experimentation.

An additional advantage lies in evaluation of new postulated designs. Currently, these designs must be fabricated and tested by human experimentation. Using a simulation model one need not physically construct the prototype. Rather, one need only describe is to the simulation model. Obviously, this approach would greatly increase the number of prototype devices which might be examined and the cost would be much reduced.

The modified Montgomery modal, discussed in Chapter 4, represents a good start toward a useful modal. One may be reasonably confident of survival times calculated from predicted cooling rates generated by this model. There is, however, some room for improvement. The major areas include: improvement of the controller, more detail in the physical modal and improved methodology for the determination of protective device thermal properties.

The most immediate need is for an improved controller. Review of the literature has shown that all controllers presently in use were designed from data collected in low rate of heat loss experiments, generally nude cold air immersions. The validation of models with those controllers has been accomplished under the same conditions. There is no reason to expect that a controller thus developed will function correctly under the conditions one encounters during cold water immersions (e.g., high rates of heat loss).

As has been demonstrated, improved model performance may be obtained by using a controller developed from data collected during cold water immersions. The controllers investigated in this study were all based on

simple deviations of temperatures st various body sites from predescribed resting temperatures for those sites. The behavior of the model, using a controller based solely on a linear regression of deviations in surface temperatures leaves much to be tried.

It was seen in Figure 111-7 that a basic problem appears to be a too slow start of the matabolic furnace. That is, the simulated metabolic rate did not increase sufficiently, compared to the rate of heat loss, until a large rectal temperature drop occurs. This leads one to believe that controllers based on rate of heat loss, rate of change of various body site temperatures or whole or partial body heat flux may be required in order to obtain improved accuracy in temperature prediction from the model. One of the reasons Gordon's model was chosen for inclusion in this study was because its controller, unlike others, used whole body heat flux in determining deviation of metabolic rate from its basal level. Unfortunately, most of the experimental work has been carried out under low rate of heat loss conditions. That work which has been performed under conditions of high rates of heat loss has not been done with controller formulation in mind. One finds that metabolic rates were not always taken or if taken were taken at long intervals. Referring to Figure 1-4 we observe a great deal of fluxation in metabolic rate as a function of time.

improved controllers can certainly be developed with existing data. More human immersion work may need to be done continuously recording metabolic and temperature data in order to formulate an accurate controller. An attempt could then be made to correlate data collected in this fashion with data from low rate of heat loss work in order to develop a controller good for all modeling dealing with conditions of heat loss.

As was discussed in Chapter 4, most of the suits which failed the paired t test were believed to fail, at least in part, because of the model's inability to simulate the flotation attitude observed in the experimentation. The Montgomery model assumes immersion to the neck. Experimentally (Figure 1-1) it was often seen that much of the trunk, legs and arms was exposed to air. Obviously, the model should and did predict higher than observed cooling rates. Therefore, to be applicable to the evaluation of suits of this type (basically abandon-ship suits) the model should be modified.

An additional problem encountered with Montgomery's model was the inability to properly describe some of the suits. The model represents

the human body is composed of a complete hand and trunk, complete arms, hands, legs and face. There is no way to distinguish between thorax and abdomen, upper and lower arm or upper and lower legs. As a consequence, protective devices which only partially covered the extremities were modeled as not covering them at all (davices such as the Medalist Ski Shorty and U-ViC Thermofloat).

It is certainly possible to model this structure in more detail. As, for example, Gordon's model does. This affords one the capability to more accurately describe the physical structure of the device to be evaluated. One would expect this to lead to much more reliable predictions from the model.

No mode: thus far reviewed has the detail in the physical system to allow adequate description of flotation attitude. The inability is inherent in the modeling philosophy adapted by all authors. Specifically, each body element has been modaled at either a cylinder or a sphere. In order to derive the differential equations describing heat conduction within cylindrical body elements, a simplifying assumption has been made: all hear transfer is radia; and uniform. To model an element partially exposed to air would necessitate consideration of longitudinal heat transfer. The derivation of model equations including this consideration would not be easy.

The changes necessary to give proper consideration to finition attitude in the model may not be nacessary. We have seen that parts of the legs, arms and trunk are exposed to air for some suits (Figure I-I). In rough sea or other open water conditions the continual washing of water over surfaces exposed to the air should result in a rate of heat loss very similar to that which would occur if the surface was continually covered by water. Under this assumption, the model's prediction of cooling rate may be very much in line with reality for these types of protection devices.

No method has been included in the modified Montgomery model for simulating flushing in wet-mode suits. Flushing is believe to do two things. First, it periodically places a range amount of cold water between suit and skin thus causing a rapid heat loss. Second, the water acts as additional insulation, once warmed, between the subject and his environment. Little is known of the dynamics of flushing and no attempt was made to include its

effects in these simulations. In extremely rough water, flushing may play a significant role in increasing the rate of heat loss.

Finally, as was pointed out in Chapter 3, a more accurate, repeatable method for detarmining device conductivity should be developed. The mothod suggested in Chapter 3 has been seen to work well. It provides a direct measure of conductivity obtained by testing the devices in the wet. Testing the materials under dry conditions is not recommended as many devices contain in their composition porous materials whose thermal properties are different in water than in air.

While may refinements may and should be made to Montgomery's model, its ability to pass a significant number of the paired t tests indicates its potential usefulness for predicting cooling rates and thus survival times. The model could, with the present modified controller, perhaps be used in screening devices prior to in vivo tests. In the long term, a modified model with improved physical definition of the body, provisions for simulating flushing and a more accurate controller would be very useful in the suit design and evaluation process. The physical system modification can be accomplished easily. The modification of the controller may require additional human experimentation.

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APPENDIX A

THE WINTON MODEL

(MODIFIED)

This model is coded in CSMP, a FORTRAN-based generalized continuous—systems simulation language. The listing includes data required to simulate a "normal" man defined as 180.4 cm in height and weighing 73.4 kg with a surface area of 1.831 m². The model is set up to simulate immersion in a full body suit with thermal conductivity of 0.499 kcal/hr·m· $^{\circ}$ C and a thickness of 0.00631 m (1/4 in.). The following table identified the major variables of the model.

MAJOR VARIABLES OF THE WINTON MODEL

Symbol	Definition	Units
тс	Temperature of the core	° C
TW	Temperature of the water	°C
TS	Temperature of skin	°C
SKDEL	Deviation of skin temperature from set point	°C
HE	Evaporative heat loss	cal/sec
HW	Heat produced by the protective device (if applicable)	cal/sec
GA	Conductance of environment	cal/sec-°C
GW	Conductance of protective device	cal/sec-°C
TE	Environmental temperatura	°C

A program listing follows with input data and output. The model is set-up to simulate a protective device 0.00631 m thick with a conductivity of 0.0499.

```
LABEL THERMOREGULATION MODEL
 * THIS RUN TESTS WATER IMMERSION RESPONSE OF MODEL
 * THERMAL ANALOG OF HODY
 TC = INTGRL (ICTC, - (GV/CC) * (TC - TS) + (HC/CC))
 TS = INTGRL (ICTS, -(GV/CS)*(TS-TC) - (GW/CS)*(TS-TW) - (HW/CS))
 TW=INTGRL (ICTW,-(GW/CW)*(TW-TS)-(GA/CW)*(TW-TE)-(HE/CW)+(HW/CW))
 * FEEDPACK STRUCTURE
 SLIM=COMPAR (33.0, TS)
 SKDEL=15-33.0
 W=LEDLAG(1400.,200., SKDEL)
 SKINFB=SGAIN*W*SLIM
 COREFB=CGAIN * TC
 TRA = TREF-SKINFB
 ERROR=TRA-COREFB
 * CONTROLLER FUNCTIONS
 GV = A FGEN (CURVEZ > ERROR)
 HSH = AFGEN (CURVE3 JERROR)
 HC = H SH
 DB1 = DERUG (3,4000.)
 DB2 = DEBJG (3,6400.)
 DB3 = DEBUG (3 , 17000.)
 * THIS CIRCUIT PROVIDES A STEP CHANGE IN TE AT 1800 SEC
 Y=STEP(1500.)
 TE= F CNSH (Y, 37.,37.,1.6)
 * DATA VALUES
 * JNITS TIME IN SEC, TEMP IN DEG C, H IN CAL/SEC, C IN CAL/DEG C
 * UNITS G.C.R IN CAL/SEC-DEG C. S IN DEG C/MM HG. V IN M/SEC
 INCOM ICTC=37.0, ICTS=33.1, ICTW=33.0
 PARAM CC=58000.0, CS=3000.0, C1=535.0
 FARAM SGAIN=0.025, CGAIN=1.7, TREF=37.1
 PARAM HE=0.0, HW=0.0, GA=13.0, GW=15.23
 AF3EN CURVE2=(-1000.,35.,-.5,35.,0.,5.,1000.,5.)
 AFGEN CURVE 3= (-1000.,18.,.6,18.,1.4,100.,1000.,100.)
 METHOD RKS
 TIMER DELT= 0.1/FINTIM=9000.0/PRDEL=100.0/OUTDEL=100.0
 PRIPLT TE(32.,42.,TC,TS,TW)
 FRITILI TE (32., 42., HC, HE)
 END
 STOP
OUTPUT VARIABLE SEQUENCE
       221006 W
SKDE L
                      SLIM
                              SKINFP TRA
                                             COREFB EPROR GV
                                                                    HSH
       ZZ1001 TC
                      221933 TS
HC
                                      Y
                                             TE
                                                     2 21 J 05 Tw
                                                                    771307
               DP3
DR1
       062
```

```
BLOCK DATA
 COMMON/ZZFDAT/F( 78)
1 /ZZHIST/KEEP/WALARM/IZODOC/IZODO1/HC
2 /ZZISTO/1( 40)
COMMON/ZZPOIN/NP(16)
 INTEGER NP/
                        40.
                               15, 27,
                                               37,
                                                                       54
      53,
             53,
                   15,
                              3, 78,
                                              0.
                                                              41
COMMON/ZZSYMB/S11( 40)
 REAL *8 511/
                                                               * TIME
                                                                      27'
1, 'TIMEDELT',' ZZDELT', 'D ELMINZZ', 'D ELNFINT', 'I MZZFINT', 'PRDEL ZZ'
                                                             ", " TW
1, PR DEOUT D', 'ELZZOUTD', 'D ELMAXZZ', 'DELXTC ', '
                                                                      77 1
                                                      TS
1,'1007ZZ10','01ZZ1003','ZZ1005ZZ','1006ICTC','
                                                             "," ICTA
                                                      JCTS
                                                                      77 '
                                                                      HW *
1,'10 C9CC ',' CGAIN ', "CS CW', ' GA ', " GW
                                                             ', ' HE
       SGAI', 'N TREF
ERRO', 'R GV
TRA ', ' W
1, 1
                       ", "C URVEZCU ", "R VE 3CORE ", "FBDB1
                                                            1, 1082
                                                                      63.
                                 HS ', "H SKDE ', "L SKINFB ', "SLIM
1,13
                        " - " H C
                                                                      TE .
1,1
                        1,14
1/
END
```

```
SUBROUTINE UPDATE
      COMMON TIME
     1,720000 DELT
                      ~ZZDELT~DEL MIN~ZZDELN~FINTIM~ZZFINT~PRDEL ~ZZPRDE
                                                     . TW
                                             . TS
     1 FOUT DEL FZ ZOUTD F DEL MAX F ZZD ELX FTC
                                                             ~ZZ1007, ZZ1001
                                                     .ZZ1009.CC
     1,221003,221005,221006,10TC ,10TS
                                              . ICTW
                                                                     . CGAIN
     1,05
              . CW
                              . GW
                                      HE
                                              - HW
                                                     SSAIN STREE
                                                                     . CURVEZ
                      ø GA
                              . DRZ
     1, CURVE3, COREFB, DB1
                                      .DB3
                                                             -HC
                                              FERROR .GV
                                                                     HSH
                                              , W
                              , TE
                                      . TRA
                                                     . Y
     1, SKDEL , SKINFB, SLIM
      COMMON/77HIST/KEEP, NALARY, 170000, 120001
      REAL
               ICTC
     1, ICTS , ICTW
      REAL *8 Z ZTIME
      EQUIVALENCE (ZZTIME, TIME )
      GO TO (39995, 39996, 39997, 39998), 170000
     SYSTEM SEGMENT OF MODEL
39995 CONTINUE
      GO TO 39999
    INITIAL SEGMENT OF MODEL
39996 CONTINUE
      50 TO 39999
    DYNAMIC SEGMENT OF MODEL
39997 CONTINUE
      SKDEL=TS-33.0
      ZZ1906=(SKDEL-ZZ1907)/200.
      w=1400.*271006+221007
      SLIM = COMP AR (33.C,TS)
      SKIN FB=SGAIN+W+ SLIM
      TRA = TREF - SKINFB
      COREFB=CGAIN*TC
      ERROR=TRA -COREFE
                      1, CURVEZ, ER FOR)
      SV=AFGEN(
      HSH = AFGE V (
                       6, CURVE 3, E RROR)
      HC=HSH
                   =-(GV/CC)*(TC-TS)+(HC/CC)
      271931
C
                              (I CTC
      T C.
                  =INTSRL
                                             ·221001
      221303
                   =-(GV/CS)*(TS-TC)-(GW/CS)*(TS-TW)-(HW/CS)
C
                   =INTGRL
                                (I CTS
                                             · ZZ 1 103
      T S
                                                          )
      Y=STEF(
                  14,1800.)
      TE=FCNSA(Y,37.,37.,1.8)
      271005
                  =-(GW/CW)*(TW-TS)-(GA/CW)*(TW-TE)~(HE/CW)+(HK/CW)
                                             , ZZ1 115
      T 94
C
                  =INTGRL
                                (1 CTW
                                                          1
      221007
                                                          )
                  =INTGRL
                                (Z Z1009
                                             .ZZ1006
      DB1=DEBUG (
                      11,3,4700.)
      DB2=DEBUS (
                      12,3,6400.)
      DB3=DEBUS (
                      13,3,17000.)
      GO TO 39999
   TERMINAL SEGMENT OF MODEL
39998 CONTINUE
```

THERMOREGULATION MODEL

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175316	37.00	i		• ÿ I	• 0 1	37.397	17.275	37, 146
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THERMOREGULATION MODEL

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APPENDIX B

THE GORDON MODEL

(UNMODIFIED)

This model is coded in standard FORTRAN and has been run on an IBM 370/168 and an IBM 3033. Run time on the IBM 3033 is about two CPU minutes for a simulated experiment of three hours length. As stated in the body of this report, the model never performed properly. The only modification appearing in the listing was made for ease of implementation. The following table lists the major variables of the model.

MAJOR VARIABLES OF THE GORDON MODEL

Symbol .	<u>Pefinition</u>	Dimension
C (N)	Heat capacitance of compartment N	kcal . °C ⁻¹
T (N)	Temperature of N	°C
F (N)	Rate of change of temperature in N	°c . h ⁻¹
HF (N)	Rate of heat flow into or from N	kcal . h ⁻¹
TC (N)	Thermal conductance between N and N + I	kcal . h ⁻ . °C ⁻
TD (N)	Conductive heat transfer between N and N+I	kcal . h
QB (N)	Basal metabolic heat production in N	kcal . h ⁻¹
Q (N)	Total metabolic heat production in N	kcal . h ⁻¹
EB (N)	Basal evaporative heat loss from N	kcal , h ⁻¹
E (N)	Total evaporative heat loss from N	kcal . h ⁻¹
BFB (N)	Basal effective blood flow to N	1. h ⁻¹
BF (N)	Total effective blood flow to N	1. h ⁻¹
BC (N)	Convective heat transfer between central blood and N	kcal . h ⁻
HC (1)	Convective and conductive heat transfer coefficient for Sagment I	kcal . m ⁻² . h ⁻¹ . °c ⁻¹
s (1)	Surface area of Segment I	m ²
HR (1)	Radiant heat transfer coeff. for Segment I	kcal . m ⁻² . h ⁻¹ . °C ⁻¹
H (1)	Total environmental heat transfer coeff. for Segment	kcal . h . ° c - 1
V	Air velocity	m . sec
TAIR	Effective environmental temperature	° C
RH	Relative humidity in environment	
TIME	Elapsed time	h
PAIR	Vapor pressure in environment	mm Hg
ITIME	Elapsed time	mîn
INT	Interval between outputs	min

Symbol 1	<u>Definition</u>	Dimension
OT	integration stap	h
P (1)	Vacor pressure table from 5 - 50°C	mm Hg
EMAX (I)	Calc, max, rate of evaporative heat loss from Sugment I	kaal . h
WORK	Total metabolic rate required by exercise	kcal h ^{r1}
PSKIN (1)	Saturated water vapor pressure at skin temp.	mm Hg
SET (N)	"Set point" or reference point for receptors	°C
ERROR (N)	Output from thermoreceptors In compartment N	'c
RATE (N)	Dynamic sensitivity of thermoreceptors in N	h
COLD (N)	Output from cold receptors in N	°C
WARM (N)	Gutput from warm receptors in N	°C
COLDS	Integrated output from skin cold receptors	°C
WARMS	integrated output from skin warm receptors	°C
SWEAT	Total efferent sweat command	kcal. h
CHILL	Total efferent shivering command	Roal. h
DILAT	Total effarent skin vasodilatation command	i. h ⁻¹
STRIC	Total efferent skin vasoconstriction command	N. C.
SKINR (I)	Fraction of all skin receptors in Segment I	N. D.
SKINS (I)	Fraction of sweating command applicable to skin of Segment	N. D.
SKINV (I)	Fraction of vasodilatation command applicable to skin of Segment 1	N. D.
SKINC (I)	Fraction of vasoconstriction command applicable to skin of Segment I	N. D.
MWORK (I)	Fraction of total work done by muscles in Segment 1	N. D.
MCHIL (I)	Fraction of total shivering occurring in muscles of Segment (N. D.
CSW	Swearing from head core	kcal . h . ° C - 1
SSW	Swammating from skin	ksal . h ⁻¹ .°C ⁻¹
CDIL	Vasodilatation from head core	1 . h . °C 1
SDIL	Vasodilatation from skin	I . h . ° C ~ 1
CCON	Vasoconstriction from head core	•c ⁻¹
SCON	Vasoconstriction from skin	•c ⁻¹
CCHIL	Shivering from head core	kcal . h . °C
SCHIL	Shivering from skin	kcal . h . • C 1
PSW	Sweating from skip and head core	kcal . h ⁻¹ .°C ⁻²

Symbol	<u>Definition</u>	Dimension
PDIL	Vasodilatation from skin and head core	1 . h - 1. ° c - 2
PCON	Vasoconstriction from skin and head core	°c ²
PCHIL	Shivering from skin and head core	°c″ ²
BULL	Factor determining temperature sensitivity of sweat gland response	°c-1

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      IMPLICIT REAL+8 (A-N.O.E)
     PIMENSION SOUT(300,14) ~ FO(300,13) ~ IMIN(10) ~ T MAX(10) ~
     *AT(34,21),IND(34),GEO(34,12),V(714),YY(34,21),EPS(34
     *>>BC(34>11)>BR(34>9>>R(34>21)>V(34>21)>VY(34>21)>AC(3
     *4),RHO(34,9),C(34,9),AK(34,9),BM(34,20),CMET(34,9),V
     *ER(34,9),TAR(34),A(34,20),B(34,20),b(34,20),E(34,20)
     ***T{34*21)**I P{t26}**TVFN(34)**U Xt34)**VECB(34)*** a(126)**
     * FNGR(34,7),EPT(34),QB(34,9),BFB(34,9),TV(34),Q3RM(3
     *4),COLD(34),FLUX(34),ERROR(34),SKINR(34),SKINS(34),S
     *KINV(34);SKINC(34);WORKM(34);CHILM(34);EB(34);FSET(3
     *4),PSKIN(*4),TO(54,21),CRATE(34),FOLD(34),EE(34),EMAX
     *(34)**NAME(34,3)*ISK(34)*IMU(34)*FW/RM(34)*FCOLD(34)
     COMMON / ENV/B C/E DR/DR/LR/R/LV/V/LVT/VT/LAC/AC/LRHG/PH
     ★O/LC/C/C/EAK/AK/LEM/BM/LCMET/CMET/CMET/LVEE/VES/LTAR/TAR/LB
     *LD/RHDB2CB2TAFF/COA/A/CO9/8/CDb/D/COE/E/LT/T/LID/NUM
     *~ISPHER ~NUMS~188~NOP~ID/LTVFN/TVEN/LUA/UA/LVELB/VFLB
     */LIA/IA/LENGR/ENGR/LEBT/EBT/LQB/QB/LBFB/BFB/LTIME/CO
     ★╭FT3ME╭EWRITE/LTV/TV/LTOT/TOTV∴TOTA/LWARM/WARM/LCOLD
     */CCLO/LF/F/LERGOR/ERROR/LSKIMR/SKIMR/LSKIMS/SKIMS/LS
     ≁KINV/SKINV/LSKIMC/XXIMC/XHORKM/WORKM/LCHILM/CHILM/LE
    ★B/EB/LHR/HR/LHC/HC/LPSKIN/PSKIN/LTC/TO/LBATE/R#TE/LT
    *OLD/TOLD/LEE/EE/LEMAX/EMAX/LIMU/IMU/LFLUX/FLUX/STOLW
    *X/TAIR~YL>EH>WORK>ECMET>CTOLD/LNAME/ANAME/LISK/ISK/L
     *STOL/CSN/SSW/PSN/CDIL/SDIL/PDIL/CCON/SCON/PCQN/CCHIL
     *~SCHIL ~PCHIL ~PULL ~ABL~PCO/LQTEN/QTEN/LFSET/FSET/FSET/LFWA
    *PM/FN4 RM/LFCOLD/FCOLD/L QTBL/GTPL/LSTOP/STOPT T/LMUS/M
    *US
     LOGICAL ASYFL
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       INPUT AND OUTPUT OF MODEL SIZE, MATERIAL, AND
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       NODAL CONFIGURATION.
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  €
     READ (1,61) TIMST, ISPHER, MUS
  61 FORMAT (F10.7,215)
     WPITE(3,31) TIMST, NUM, ISPHER, MUS
  31 FORMAT("1", T20, "EXPERIMENT DURATION =",F10,7," HOURS"
    *//TZO, TOYAL NUMBER OF BODY FLEMENTS EQUALS : 1,15//TZO
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***TOTAL NUMBER OF SPHERICAL BLEMENTS EQUALS :***15//T20
     DATEXTREMITIES START WITH BODY ELEMENT : 1215//)
      PFAD(1,32)(IA(M),M=1,34)
   32 FORMAT (4012)
      READ (1,33) (10(M), H=1,34)
   33 FORMAT (619)
      NUNTER 55 OC
      IF (IACM) .NE. C) NOF #4
   25 CONTINUE
      READ(1,62)(UACM),M=1,34)
      READ(1,133)(IMU(M),4=1,34)
  133 FORMAT (1615)
      READ(1,35)((ANAME(M,K),K=1,3),M=1,34)
   35 FORMAT (18A4)
      WRITE(3,34)(M,IA(M),ID(M),ANAME(M,I),ANAME(M,2),ANAME(
     *Mょう) >UA (M) > I MU ( N) > M#1 > ふんご
   34 FURMAT (TOOM "FLEMENT SYMMETRICAL
                                          NO DE ". 1D X. "ANATO".
     # TO YEAR.
                 COHE (KCAL/HR/DEG)
                                       MISCLE'/TEE,'NO.
     # <sup>8</sup>
        ELEMENT
                    DISTRIBUTION
                                       NAME
     *13Xa*DISTRIBUTION*//34(T22aI2aT33aI2aT4(aI10a157a3A4aT
     *76, F5, 2, T96, 34/))
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        INPUT AND OUTPUT OF GEOMETRIC SIZES, BASAL BOSY
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        PARAMETERS, AND BOUNDRY CONDITIONS.
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C
  CALL SEOMRG (GEO, NUM, 12)
     WRITE(3,66)
  66 FORMAT("1",T20,"8C(11) = CONVECTIVE H.T. COEFF. (CALT,
     *'/SQ.CM/HR/DEC)'/T20,'EES = EMISSIVITY*SHAPE~FACTOP ',
     *'(N.D.)'/TZO','BC(3) = TEMPERATURE OF CONVELTIVE GNVI',
     **RONMER1 (DEG CENTIGRADE)*/T20,*BC(4) = TFMPER4TURE */
     **OF RADIATIVE ENVIRONMENT (DEG CENTIGRADE)*/ *20, **CC(*,
     **7) = INITIAL TEMPERATURE CHANGE PERMISSIBLE AT ELD ',
     **OF ITERATION (DEG CENTIGRADE) *//T7, *ELEMENT
     *11)
            FPS
                     BC(3)
                               Bc (4)
                                           BC(7)1)
     00 20 M=1.NOP
     IF (ASYFL (M) > GO TO 2"
     READ(1,67)BC(M,11),EPS(M),BC(R,3);BC(M,4),BC(G,7)
  67 FORMAT (4F10.5,1), F10.9)
     WRITE(3,64) ANAME(M, i), ANAME(M, 2), ANAME(M, 3), PC(M, 11), E
     *PS(M)#BC(M#3)#PC(M#4)#BE(M#7)
  64 FORMAT (1X,3A4,1X,5F10.5)
     8C(M,3)=BC(M,3)+273.16
     BC(M,4)=BC(M,4)>273.16
  50 CONTINUE
     WRITE(3,65)
  65 FORMAT (///T10,*RHO = DENSITY (GM/CUPIC CM)*/T10,*C =*,

→ SPECTFIC HEAT (CAL/CM/CCG) 1/T10. 10K = CONDUCTIVE H'A
    **.T. COECFICIENT (CAL/UM/HK/DEG)*/FID>*UMET = BASAL *>
    **METABOLISM (CAL/OF)*/T10,*VCB = BASAL PLOCO FLOW (M*,
```

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*'L/HR)'///TO,'ELEMENT MATE NO.
                                           ~ XS F ~ " 0 " ~ X S F ~ " CH R --
     * "AK ", 10x, "CMET", 10x, "VEB")
      RHOCPV = 0.6
      DO 12 N=1,NOF
      U \wedge (M) = U \wedge (M) + 1000.
      XF(ASYFL(M)) GO TO 12
      CALL DEFSRG (M/LMAT/NODE/12)
      IND (M) = NODE
      ISK(M)=LMAT
      SIGEP=EPS(M) *4. 2756E-9
      HC(M,2) =SIGEF + AC(H)
      DO 11 J=1,LMAT
      READ (1, 82) THO(M, J), C(M, J), XX, (M, J), CHET (M, J), VEB (M, J)
      RHOCPV=RHOCPV+RHO(M,J) + C(M,J) + VT(M,J)
   62 FORMAT (SF12.5)
      EFB(H, J)=VEB(M, J)
      GB(M,))=CMET(M,J)
      WT=WT+RHO(M,J)*VT(M,J)/1000.001
   11 WRITE(3,63)ANAME(M,1),ANAME(M,2),ANAME(M,3),J,RHO(M,J)
     * C (M , 3 ) , AK (M , 3 ) , CM ET (M , 3 ) , V E E (M , 3 )
   53 FORMAT (1X,3A4,4X,12,2X,5(1X,F12.5,1X))
      IF (M.ST.ISPHER) GO TO 10
      ABRER(MINODE)+DR(MILMAT)
      ABRR=R (MANODE)+DR (MALMAY)/2.
      BC(M,9)=+DR(M,LMAT)/AK(M,LMAT)/AC(M)*APR*ABR/APRR/APRR
      60 TO 12
   10 ACC=AC(M)/R(M,NODE)
      BC(Mag)=-DR(MalMAT)/(R(Manode)+DR(MalMAT)/2.)/ABC/AK(M
     */LMAT)
   12 CONTINUE
      WRITE(3,74) WT
   74 FORMAT(///T2C, THE TOTAL BODY WEIGHT'/T20, EQUALS ==> "
     40F10.500 (KG)1//)
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  <> CALCULATION OF COEFFICIENTS FOR ITERATIVE ROUTINE.
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C
      CALL COEFRG
      READ (1,102) ( EB ( IM ), EM=1, NUMS )
  102 FORMAT (1665.0)
      READ (1, 150) TAIP, VL, RH, WORK, DINT
  150 FORMAT (3F10.5, F20.5, F10.8, 212)
      IM=0
      DO 302 M=1.NOP
      TELASYFL(M)) GO TO 302
      IM= 1 M > 1
      B((M,8)=RH
      BC(M,53=DINT
      BE(M,1)=BC(M,11)*AC(M)
  BURITHOD SCE
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       STEADY STATE TEMPERATURE DISTRIBUTION AND HODY
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       PARAMETERS ARE CALCULATED AND PRINTED.
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  C
     CALL STEPRG(YY,Y)
     WRITE(3,73)
  73 FORMAT (//1x, ', THE ARTERIAL BLOOD TEMPERATURES ARE A',
    *'S FOLLOWS: '//)
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          INPUT AND OUTPUT OF CONTROL COEFFICIENTS
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     IM=D
     DO 15 M=1.NOP
     IF(ASYFL(M)) GO TO 15
     IM= IM+ 1
     LSK=ISK (M)
     L=IND(M)+1
     FSET(IM)=FLUX(IM)
     DO 5 V=1.L
   5 TO(M,V)=T(M,N)
     TOLD(IM) = T(M,L-1)
     VB=VB+VT(M>ESK)
     AT(1,IM) = TAR(M) - 273.16
  15 CONTINUE
     CTOLD=T(1,1)
     WRITE(3,62)(AT(1,1M),1M=1,NUMS)
     READ(1,103) STOPIT
     WRITE(3,24) STOPIT
  24 FORMAT(//T20, SUCCESSIVE ITERATIVE TEST EQUALS ==> ", F5
    * 4)
 103 FORMAT (16F5.4)
     RFAD(1,103)(SKINR(IM),IM=1,NUMS)
     READ (1,103) (SKINS(IM), IM=1, NUMS)
     READ (1, 103) (SKINV(I4), IM=1, VUMS)
     READ(1,103) (SKINC(IM), IM=1, NUMS)
     READ (1, 103) (WORKM(IM), IM=1, NUMS)
     READ (1, 103) (CHILM(IM), IM=1, NUMS)
     WRITE(3,104)
 174 FORMAT ('1'//' ER=CAL/HF'/' FLUX=CAL/HR/SQ.CM 1//T/O.
    **FNVIRONMENTAL COEFF.*, 130, *WEIGHTING COEFFICIENTS*//
                       SKINR SKINS
    *6X, *ELEMENT
                                    SKINV SKINC
    *'KM CHILM', 10x, 'fLUX', 11x, 'EB')
     IM = 0
     DO 135 M=1,NOP
     IF(ASYFL(M)) CO TO 175
     WRITE(3,1(17) ANAME(M,1), ANAME(M,2), ANAME(M,3),SKINR(IM)
```

```
*, SKINS (IM), SKIN V(IM), SKINC (IM), WORKM(IM), CHILM (IM), FLU
     *X(IM), EB(IM)
  135 CONTINUE
  107 FORMAT (1X,3A4,8X,6F7.4,E20.8,F7.0)
      READ (1, 11()) C SW, SSW, PSW, CDIL, SDIL, PDIL, CCON, S CON, PCON, C
     *CHIL,SCHIL,PCHIL,BULL,ABL,QTEN,QTBL,PCO
  110 FORMAT (6F9.3)
      WRITE(3,112)CSW,SSW,PSW,CDIL,SDIL,PDIL,CCON,SCON,PCOM,
     *CCHIL, SCHIL, PCHIL, BULL, ABL, GTEN, QTBL, PCO
  112 FORMAT (15(/),1x,T20, CONTROLLER COEFFICIENTS
                                                     1/// T25,
     *'CSW=',F9.4,' KCAL/HR/DEG'/T?5,'SSW=',F9.4,' SQ.M'/T
           PSW="2F9.42" LITER-SQ.M/KCAL"/T252"CDIL="2F9.4
     *,' LITER/HR/DEG'/T25,'SDIL=",FQ.4,' LITER-SQ.M/KCA',
     *'L'/T25,'PDIL=',f9.4,' LITER-SQ.M/KCAL'/T25,'CCON=",F
             1/DEG'/T25,'SCON=',F',4,' LITER-SQ,M/KCAL'/T25
     *, *PCON = *, F9.4, * LITER-SQ.M/KCAL*/T25, *CCHIL =*, F9.4, 2X
     *,'KCAL/HR/DEG'/T25,'SCHIL=',F9.4,' SQ.M'/T25,'PCHIL='
     *,F9.4,' KCAL/HR/SQ.DEG'/T25,"BULL=',F9.4,' 1/DEG'/T2
     *5, ABL= 1, F9.3, LITER/DEG/HR1/T25, TQTEN= 1, F9.4, 11,
     *'/DEG'/T25,'GTBL=',F9.4,' 1/DEG'/T25,'FC0=',F9.4,' M',
     **L/CAL*///)
      CSW=CSW *1000.
      SSW=SSW *10000.
      PSW= PSW * 10000.
      CDIL = CDIL + 1000.
      SDIL = SDIL * 10 000.
      PDIL =PDIL *10000.
      SCON=SCON+10000.
      PCON =PCON *10000.
      CCHIL=CCHIL +1000.
      SCHIL=SCHIL * 10000.
      PCHIL=PCHIL * 1000.
      ABL = ABL +1000.
      50 TO 44
  90 CONTINUE
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              INPUT AND OUTPUT OF TRANSIENT
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             BOUNDRY CONDITIONS.
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  READ(1, 153) TAIP, VL, RH, WORK, DINT, LBC, IDA
     DO 202 M=1,NOP
      IF(ASYFL(M)) GO TO 202
      IF(IRC.NE.1) GO TO 22
      IF(IDA.EQ.0) GO TO 23
      READ(1,13)EPS(M),BC(M,3),RC(M,11),BC(M,8),BC(M,10)
  13 FORMAT (F5.4, F5.2, F8.4, F6.4, F8.4)
     BC(M,2)=EPS(M)*AC(M)*4.8756E-9
     BC(M,3)=BC(M,3)+273.16
     60 TO 23
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22 BC(M,8)=RH
     BC(M,3)=TAIR+273.16
     BC(M,10) = VL
  23 BC(M,7) =1,000000000001
     BC(M,5) =DINT
     BC(M,4)=BC(M,3)
     BC(M_21) = BC(M_211) * AC(M) * (BC(M_210)/_10) * *0.5
 232 CONTINUE
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  CALL STEPRG(YY,Y)
     TIME=TIME+BC(1,5)
  44 IQ=IQ+1
     IF(TIME.GE.TIMST) GO TO 82
     ETIME = TIME * 60.00001
     TPOOL= TARP-273.16
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             OUTPUT OF TRANSIENT PHYSICLOGICAL
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             PARAMETERS FOR EACH TIME STEP
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C
     WRITE(3,40) ETIME, TPOOL, ECMET
  40 FORMAT (//1x, "TIME EQUALS ", F5.1," MINUTES "/ 1x, "ARTE
    **RIAL POOL TEMPERATURE EQUALS ',F6.3,' DEG'/1X, 'RE'
    *'SPIRATORY LOSSES EQUALS ',-3PF6.3,' KCAL/HR'/)
     WRITE(3,42)
  42 FORMAT (8Y, ELEM
                     TCORE TSKIN
                                  INBAL
                                         RBEC
                                              BLHTR
                                              QCC
    ** MET
              CONV
                     RAD
                           EVAP
                                 SBF
                                       TBF
    * TAR
            FLUX
                   FOT
                          FOOL ')
     SOUT (JQ.1) = ETIME
     SOUT(IQ,13) = 0.0
     C. G=T3W3
     SOUT (IQ.12) = 0.0
     SOUT (IQ.3) = F CMET
     SOUT(IQ.4) = E CMET
     SOUT (10,6)=0.0
     SQUT(1Q,5)=0.0
     SOUT (IQ, 11) = 0.0
     SOUT (IQ, 12) = CO
     EO(10,1)=').
     EO(IQ \times 2) = 0.
     EO(10.3)=0.
     E0(19.4)=0.
     EC(10.5)=0.
     E0(10,6)=0.
```

```
EO(10,7)=().
     EO(19,8)=0.
     EC(10,9) = ECMET/1000.
     EO(IQ, 10)=TAIR
     EO(10.11)=VL
     EO(10,12)=RH
     EO(10,13)=WORK/1000.
     CTOLD=T(1,1)
     IM=7
     00 30 M=1.NOP
     IF(ASYFL(M)) GO TO 30
    L=IND(M)+1
    LSTOP=IND(M)
     IM = IM + 1
     TCLD(IM)=T(M,L-1)
     ATAR=TAR(M)-273.16
     QCC=UA(M)*(TARP-TAR(M))
     EO(10,1)=FO(10,1)+EBT(N)/1000.001
     EO(IQ, Z) = EO(IQ, Z) + ENGR(M, Z) / 1000.001
     EO(19.3) = EO(19.3) + ENGR(M.1)/1900.901
     EO(10,4)=00(10,4)+ENGR(M,2)/1000.001
    EO(14,5) = EO(10,5) + ENGR(M,3) / 1000.001
     EO(IQ,6)=EO(IQ,6)+ENGR(M,4)/1000.001
     EC(IQ, 7) = EU(IQ, 7) + ENGR(M,5) / 1000.001
    FO(IQ,8)=EO(IQ,8)+ENGR(M,6)/1000.001
    AT(IM_{e}11)=T(M_{e}L)-273.16
    AT (IM, 2) = T(M, 1) - 273.16
    LSK=ISK(M)
    SOUT(10,6)=SOUT(10,6)+(T(M,L)-273.16)*VT(M,LSK)/VR
    EWET=EWET+EE (IM)/EMAX(IM)*AC(M)/TOTA/1.000001EU4
    SOUT (IQ . 4) = SOUT (IQ . 4) + EE (IM)
    TREF=D.O
    SKEF=0.0
    1=1
    CALL DEFRG(M, IMIN, IAAX, LYAT, NODE)
    DO 801 N=1, LSTOP
    IF(N.ST.IMAX(J)) J=J+1
    AQTEN=BM(M,N)**(QTFN/QTBL)
    (L,M)TV\setminus (N,M)V \times (L,M)E3V = 2V
    CM = CMET(M_J) *V(M_N)/VT(M_J)
    IF(J.NE.LSK) 60 TO 4
    SKBF = SKPF+VS *AQTEN
  4 SOUT (IQ. 3) = SOUT (IQ. 3) + CM*PM (M.N.)
831 TBLF=TBLF+VS *ARTEN
    SOUT (IQ, 11) = SOUT (IQ, 11) + SKBF /6 0000 .01
    FOOL = FCOLD(IN) * SKINR(IM)
    FOT=FAARM(IM) + SKINR(IM)
    WRITE(3,43) ANAME(M,1), ANAME(M,2), ANAME(M,3), AT(IM,2), A
   *T(IM,11),EBT(M),ENGR(M,1),ENGR(M,2),ENGR(M,3),ENGR(M,4
   *),FNGR(M,5),ENGR(M,6),SKRF,TPLF,QCC,ATAR,FLUX(IM),FOT,
   *FOOL
43 FORMAT (1X,3A4, PP2F7, 2, -3P10F7, 2, OP1F7, 2, OP3F7, 2)
33 CONTINUE
    IF(IDA.EQ.0) GO TO 36
    WRITE(3,37)
```

```
57 FORMAT (/1x, 125('x')/T30, '** WARNING
                                       **'/T2 (, 'BOUND',
    *'RY CONDITIONS HAVE BEEN CHANGED'/7X, "ELEM"
                                               EPS (M) .
    *2 X, 'BC (M,11)
                   BC(M,R)
                            BC(M, 10) 1/)
     DO 38 M=1,NOP
     IF(ASYFL(M)) GO TO 38
     BCDUM=BC(M,3)-273.16
     WPITE(3,39) ANAME(M,1), ANAME(M,2), ANAME(M,3), EPS(M),8CD
    *UM,BC(M,11),BC(M,8),BC(M,10)
  39 FORMAT(1%,3A4,2X,F5.4,4X,F5.2,5X,F8.4,3%,F6.4,4%,F8.4)
  38 CONTINUE
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      STORAGE OF PHYSIOLOGICAL DATA INTO SUMMARY ARRAYS
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  C
  36 EC(IQ, 5) = EO(IQ, 5) + EO(IQ, 9)
     SOUT (IQ,4)=SOUT (IQ,4)/TOTA/1000.001
     SOUT (IQ, 3) = SOUT (IQ, 3) / TOTA / 1000.001
     SOUT(IQ,2)=EG(IQ,3)/TOTA
     SOUT (10,13) = (SOUT (10,3) - ECMET/TOTA/1000.001- SOUT (10,2)
    *)/(TARP-273.16-SOUT(10.6))
     SOUT (10,5)=E0(10,2)*1000.001/RHOCPV
     SOUT (IQ,10) = T(7,2) - 273.16
     SOUT(IQ,9)=T(2,2)-273.16
     SOUT(IQ\times8)=TARP-273.16
     SOUT (10.7) = T(1.2) - 273.16
     SOUT (IQ, 12) = SOUT (IQ, 12) /60000.01
     SOUT (IQ, 14) = 100. *EWET
     60 TO SO
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  <> OUTPUT OF SUMMARY ARRAYS FOR TRANSIENT RESPONSE
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  C
  32 K=60
     L=1
     LX=IQ-1
  77 WRITE(3,75)
     IF(LX.LT.60) K=K-60+LX
  75 FORMAT ('1', TIME HFLOW
                              HP
                                     E۷
                                           TP
                                                TS
                                                PWET ')
                                    CO
    * TH
                  TR
                             S3 "
                                          COND
             TO
                       TM
     WRITE(3,76)((SOUT(II,JJ),JJ=1,14),II=L,K)
  76 FORMAT (1X,F5.1,3F7.2,6F6.2,2F7.3,2F7.2)
     IF(LX.LE.60) GO TO 78
     LX=LX-60
     L=L+60
     K = K + 60
     60 TO 77
  78 K=60
     L=1
```

```
LX=IQ-1
   87 WRITE(3,85)
      IF(LX.LT.60) K=K-60+LX
   85 FORMAT ("1", "ALL UNITS IN KCAL/HR EXCEPT TIME = MIN.,",
     ** TAIR = DEG-C, VEL = M/SEC, AND RH IS NON-DIMENSION*,
     * AL " // TX, TIME
                       INBAL
                                 BEC
                                          RBFC
          MET
                   CONV
                             RAD
                                    EVAP
                                             RESP
                                                      TAT"
                             WORK 1)
             VEL
                       RH
      WRITE(3,86)(SOUT(II,1),(EO(LI,JJ),JJ=1,13),II=L,K)
   86 FORMAT (1X, F5, 1, 13F9, 3)
      IF(LX.LE.60) GO TO 88
      LX=LX-60
      L=L+60
      K=K+60
      GO TO 87
   88 CONTINUE
      STOP
      ENU
C
 C <>
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C
      BLOCK DATA
     IMPLICIT REAL*8 (A-H,0-Z)
     DIMENSION BC (34,11), DR (34,9), R (34,21), V(34,27), VT (34,9
     *)/AC(34)/RHO(34/9)/C(34/9)/AK(34/9)/BM(34/23)/CMET(34/
     *9),VE3(34,9),TAP(34),A(34,20),B(34,20),D(34,20),E(34,2
     *O)aT(34,21)aID(126)aTVEN(34)aUA(34)aVELB(34)aIA(126)aE
     *NGR(34,7),EBT(34),QB(34,9),PF3(34,9),TV(34),WARM(34),C
     *OLD(34),FLUX(34),ERROR(34),SKINR(34),SKINS(34),SKINV(3
     *4), SKI NC (34), WORKM (34), CHILM (34), FF (34), FSET (34), FSKIN
    #(34),TO(34,21),RATE(34),TOLD(34),EF(34),EMAX(34),ANAMF
     *(34,3),ISK(34),IMU(34),FWARM(34),FCOLD(34)
     COMMON/ENVZEC/LDR/DR/LR/R/LV/V/LVT/VT/LAC/AC/LRHO/RHO/
     *LC/C/LAK/AK/LBM/BM/LCMET/CMFT/LVEB/VEB/LTAR/TAR/LBLD/R
     *HOB,CB,TARP/COA/A/COP/B/COD/V/COE/F/LT/T/LID/NUM,ISPHE
     *R,NUMS,ISS,NOP,ID/LTVEN/TVEN/LUA/UA/LVELB/VELB/LIA/IA/
     *LENGR/ENGR/LEBT/EBT/LQB/QB/L@FB/BFB/LTIME/CO.ETIME.FWR
    ≠ITE/LT∀/TV/LTOT/TOTV,TOTA/L%ARM/WARM/LCOLD/CGLD/LF/F/F/L
    *ERROR/ERROR/LSKINR/SKINR/LSKINS/SKINS/LSKINV/SKINV/LSK
    *INC/SKINC/LWORKM/WORKM/LCHILM/CHILM/LEB/EB/LHR/HR/LHC/
```

*HC/LPSKIN/PSKIN/LTO/TO/LRATE/RATE/LTOLD/TOLD/LEE/EE/LE *MAX/EMAX/LIMU/IMU/LFLUX/SLUX/STOLWX/TAIP~VL/RH/WORK/FC *MET,CTQLD/LNAME/ANAME/LISK/ISK/LSTQL/CSW,SSd,PSW,CDIL, *SDIL,PDIL,ECON,SCON,PCON,CCHIL,SCKIL,PCHIL,3ULL,ABL,PC *O/LQTEN/QTEN/LFSET/FSET/LFWARW/FWARM/LFCOLD/FCOLD/LQTB *L/QTBL/LSTOP/STOPIT/LMUS/MUS LOGICAL ASYFL DATA CSW/SSW/PSW/CDIL/SDIL/PDIL/CCON/SCON/PCON/CCHIL/S +chil,pchil,bull,abl,pc0/12*0.0.*10.0003..1.6/,aten/10 *00000./,QTBL/1000000./,STOPIT/.001/,f/714*309./,IA/1,2 *,4,3,6,5,8,7₆10,9,116*0/,BC/374*0.0/,ENGP/238*0.0/,EAT */34*0.0/,NUM,ISPHER,ISS,ID/34,1,1,10*1111,116*0/,CO,FT *IME,EWRITE/300000.,2*0.0/,TAR/34*309.93637/,RH08,09/1. *08.0.92/,TARP/309.93637/,T0/714*309.0/,TAIR,VL,RH,WORK */29_45_0.10_0.3_0.0/_RATE/34*0.0/_UA/34*0./_SKINP/34*0 *.O/,SKINS/34*0.0/,SKINV/34*0.0/,SKINC/34*0.0/,WORKM/34 **C.O/,CH1LM/34*J.O/,FLUX/34*O.O/,FSET/34*D.O/,RF8/136*

C C <> **< >** C <> <> C <> <> <> C <> C <> () C <> <> C <> <>

SUBROUTINE SCONT
IMPLICIT REAL+8(A-H,O-Z)

END

DIMENSION TS (11),P(11)

*0.//QB/136*0.//6M/630*1.0/

bimensionec(1),vt(1),ac(1),bm(1),cmet(1),veb(1),t(1),i *D(1),engr(1),qb(1),bfb(1),imu(1),ak(1),r(1),fset(1),fw *ARM(1),fcold(1),warm(1),cold(1),flux(1),erp)r(1),skinr *(1),skins(1),skinv(1),skinc(1),workm(1),chilm(1),eb(1) *,pskin(1),to(1),rate(1),told(1),ee(1),emax(1),aname(1) *,isk(1)

COMMON/FNV/BC/LVT/VT/LAC/AC/LBM/BM/LCMET/CMET/LVEB/VEB
*/LT/T/LAK/AK/LID/NUM,ISPHER,NUMS,ISS,NOP,ID/LQB/QB/LBF
*B/BFB/LIMU/IMU/LTIME/CO.ETIME,EWRITE/LR/R/LW ARM/WARM/L
*COLD/COLD/LF/F/LERROR/ERROR/LSKINR/SKINP/LSKINS/SKINS/
*LSKINV/SKINV/LSKINC/SKINC/LWORKM/WORKM/LCHILM/CHILM/LE
*B/EB/LPSKIN/PSKIN/LFLUX/FLUX/LTO/TO/LRATE/RATE/LTOLD/T
*OLD/LEE/EE/LEMAX/EMAX/STOLWX/TAIR,VL,RP,WORC,ECMET,CTO
*LD/LNAME/ANAME/LISK/ISK/LSTOL/CSW,SSW,PSW,CDIL,SDIL,PD
*IL,CCON,SCON,PCON,CCHIL,SCHIL,PCHIL,BULL,ABL,PCO/LQTEN
*/QTEN/LFSET/FSET/LFWARM/FWARM/LFCOLD/FCDLD/_GTBL/QTBL/
*LMUS/YUS

LOSICAL ASYFL

DATA T\$/0.0,5.,10.,15.,20.,25.,30.,35.,40.,45.,50./
DATA P/4.579,6.543,9.209,12.788,17.535,23.756,31.824,4
*2.175,55.324,71.88,92.51/

```
VS/ZI ATAD
    RATES=().0
    AG0 = 1.
    SUMT=3.0
    WARMS= 0.0
    FWARMS = 0.0
    COLD S= 0.0
    FCOLDS = 0.0
    CWARM= ().3
    CCOLD=0.0
    CRATE= (T(1)-CTOLD)/BC(137)
    IF(WORK.GT.71732.) SO TO 105
    WORKI=0.0
    60 TO 106
135 WORK 1= (WORK - 71732.) +0.78
106 IN=0
301 DO 302 M=1,NGP
    IF (ASYFL(M)) GO TO 302
    I 4M I = P' [
    CALL DEFSRG(M,LMAT,NODE,IZ)
    L=M+NUM * (NODE-1)
    LL=M+VUM+(NCDE-2)
    FLUX(IM)=(T(LL)-T(L))*AK(M+NUM*(LMAT-1))/(R(LL)-R(L))
    RATE (IM) = (T(L) - TOLD(IM))/3c(M+NUM+4)
    RATES=RATES+RATE(IM)*SKINR(IM)
    00 305 N=1, NODE
    JN=M+VUM*(N-1)
306 BM(JN) = 2.**((T(JN)-TO(JN))/QTEN)
    WARM(IM)=0.0
    FUARM(IM)=0.0
    COLD(IM)=(.C
    FCOLD(IM)=0.0
    MSK=M+NUM+(NODE-1)
316 ERPOR(IM)=T(MSK)-TO(MSK)
    FERROR = FLUX (IM) - FSET (IM)
    IF(ERROR(IM).GT.5.3) GO TO 3(4
    1f(ERROR(IM).EQ.0.0) GO TO 307
    COLD (IM) =-ERROR (IM)
    COLDS=COLDS+COLD(III)*SKINR(IM)
    SUMT=SUMT+(TO(MSK)-273.16)*SKINR(IM)
    60 TO 367
334 WARM(IM) = EPROR(IM)
    WARMS=WARMS+WARM(IM) + SKINR(IM)
307 IF (FERROR.GT.P.C) GO TO 308
    IF (FERROR.FO.C.C) GO TO 302
    FCOLD(IM)=-FERROR
    FCOLDS = FCOLDS+FCOLD(IM) *SKINR(IM)
    60 TO 302
308 FWARM(IM)=FERROR
    FWARMS = FWARMS+FWARM(IM) *SKINR(IM)
332 CONTINUE
    TCOLD=COLDS-SUMT+28.5
    CERROR = T(1) - TO(1)
    IF (CERROR.LT.O.O) GO TO 303
```

```
CWARM# CERROR
     FCCOLD = -CERROR
     60 TO 305
333 CCOLD=-CERROR
     FCCOLD = -CERROR
335 SWEAT= CSW+CWARM+SSW+FWARMS
     DILAT=CDIL+CWARM+SDIL+FWARMS
     STRIC= C CON+C COL D+SCON+F COLDS+PCCN+COLDS
     IF(TCDLD.LT.C.0) AGJ=0.
     CHILL= CCHIL * FCCOLD+SCHIL *FCOLDS+PCHIL *TroLD* AGO
     IF(SWEAT.GE.O.O) GO TO 312
     SWEAT= 0.0
312 IF(DILAT.GE. C. 0) GO TO 314
     O. GETALIG
314 IF(STRIC.GE.C.O) GO TO 316
     STRIC=0.0
316 IF(CHILL.GE.C.D) GO TO 318
    CHILL=9.0
318 IM=0
430 00 401 M=1.NOP
    IF (ASYFL(M)) GO TO 471
    IM=IM+1
    L=ISK(M)
    cus=0.0
    IF (M.SE.MUS) CUS = (ABL * COLDS + PDIL * FCOLDS) * CHIL M(IM)
    CALL MUSCLE (M, INUM, KA, KB, KC)
    IFCINUM.LE.D) GO TO 503
    GO TO (510,520,536), INUM
530 NC=M+VUM+KC
    NB=W+N N₩*KB
    NA=M+VUM*KA
    VM=VT(NC)+VT(ND)+VT(NA)
    CMET(NC)=RB(NC)+(WORKM(IM)+WORKI+CHILM(IM)+CHILL)+VT(N
    VEB(NC) = BFB(NC) + (WORKM(IM) * WORKI+CHILM(IM) * CPILL) * V T(N
   *C)/VM*PC0-CUS
    IF(VEB(NC).LE.O.005*BFP(NC)) VEP(NC)=.005*PFP(NC)
    GO TO 525
520 NB=M+VUM+KD
    NA=M+VUM+K4
    VM=VT(N()+VT(1.A)
525 CMFT(VB)=OB(NP)+(WORKM(IM)+WORKI+CHILM(IM)+CHILL)+VT(N
   *P3/VM
    VEB(N3) = BFR(NB) + (workm(IM) + work1+chit4 (IM) +chitt) + VT(N
   *B)/VM* PCC-CUS
    IF(VER(NB).Lf.0.005*BFB(NB)) VER(NB)=.005*BFR(NB)
    60 TO 515
510 NA=M+YUM*KA
    VM=VT(NA)
515 CMET(VA)=QB(NA)+(WGRKM(IM)+WGRKI+CHILM(IM)+CHILL)+VT(N
   MVICAR
    VEB(NA)=PFB(NA)+(WORKM(IM)+WORKI+CHILM(IM)+CHILL)+VT(N
   *A)/VM*PCO-CUS
    IF(VEB(NA)_LF.0.305+BFH(NA)) VER(NA)=.005+BFH(NA)
```

```
533 NJ=M+VUM+(L-1)
    VEB(NJ)=BFB(NJ)-STRIC*SKINC(IM)+DILAT*SKINV(IM)
    IF(VEB(NJ)=.005+PFR(NJ) VEB(NJ)=.005+PFR(NJ)
5D1 &E(IM) = (EB(IM)+SKINS(IM)+SWEAT)+2.0++(ERROR(IM)/BULL)
    CALL DEFSEG(MALMATA VODE 212)
    T7=BC(M+NUM+2)-273.16
    CALL TERFRO(TS,P,TZ,PAIR,11)
    PAIR = PAIR + HC (M+ NUM+7)
    NU-NODE +1
    TSK=T(M+NUM+(NU-1))-273.16
    CALL TERPRO(TS,P,T9K,PSKIN(IM),11)
    EMAX(IM)=(FSKIN(IM)-PAIR)+2.2+PC(M)
    IF CEECIMOLLE LEMAX (14) ) GO TO 403
    EU(IM) = EMAX(IM)
433 PC(M+VHM+5) = EE(IM) +T(M+ NUM+(NU-1))
401 CONTINUE
    50M6T= (71732.+WORKI/.78+CHILL) +0.0023+644.0-PAIR)
    CMET(3)=QB(3)-ECMEY*.2
    CMST(38)=QB(38)-FCMET*.1
    CMET (71) = QB (71) - FCMET + . 7
    VEB(5) = CO + BFF(5)
    VER(6) = BFB(6) - PSW*FWARMS
    RETURN
    END
    SUPROUTINE COEFFG
    IMPLICIT REAL+9(A-H,O-Z)
    DIMENSION IMIN(10), IMAX(10)
    DIMENSION DR (1) - R (1) - VT (1) - TAR (1) - R HO (1) - C (1) - AK (1) - PM (1) -
   $CMET(1),VEP(1),A(1),P(1),P(1),D(1),E(1),ID(1),QB(1),REB(1)
    COMMON /LDR/D R/L R/R/LVT/ VT/LR HO/RHO/LC/C/LAK/ AK/LBM/PM
    COMMON/LCMET/CMET/EVER/VEB/LTAR/TAR/LBLD/RHOD, CB, TARP/COA/A/COR/B
    COMMON/COD/D/COE/E/LID/NUM/ISPHER/NUMS/XSS/NOP/ID
    COMMON / LGB/GB/LRF8/3FS
    COMMONIAL RIEN /GIEN
    COMMON/LUTEL/RITEL
    LOGICAL ASYFL
    160=1
    15P=1
    DO 99 1=1.NOP
    IF (I.ST.ISPHER) ISP=?
    ICEN=1
    IF(ASYFL(I)) GO TO 9.
    CALL DEFRECT, IMIN, IMAX, LMAT, NODE)
    K=1
    If (IMIN (K).EG.IMAX(K)) K=2
    60 TO (23,24),K
24 ICEN=2
23 BB=AK(I+NUM+(K-1))/DR(I+NUM+(K-1))/4.
    DO 80 N=2.NODE
    MP=I+VUM+K
    MM=1+VU*+(K-1)
    MN=1+VUM+(N-1)
    IF (IMIN(K).EQ.IMAX(K)) GO TO 14
    IF (N.EQ.IMIN(K)) GO TO 5
```

```
IF (N.L.Y. TMAX(K)) GO TO 5
  4 TFLOWERS
    IF (N.EQ.NOOE) GO TO 12
    AA FRB
    RP=DR(MM)+DR(MP)
    BB=(AK(MM)*DR(MM)+AK(MP)*DR(MP))/RP/RP/?。
    ICO=5
    50 TO 40
 & FF(N.EQ.2) GO TO 16
    AA=BB
   BB=AK(MM)/DR(MM)/4.
   GO TO 40
 98=AA 8
   60 TO 40
19 IFLOW=2
12 AA=88
   GO TO (46,47), ISP
46 BB=AK(MM)/DR(MM)/2.
    GO TO 48
47 BBT=R(MN)/DR(NM)
   88=AK(MM)*(BRT+.5)/(2.*8BT+1.)/DR(MM)
48 GO TO (40,17), IFLOW
14 IF(N.EQ.NODE) GO TO 19
   RP = DR(MM) + DR(MP)
   BB=(AK(MM)*DR(MM)+AK(MP)*DR(MP))/RP/RP/2.
   IG0=2
   60 TO 17
16 BB=AK(MM)/DR(MM)/4.
17 RP=DR(MM)+DR(I+NUM*(K-2))
   AA=(AK(MM)*DR(MM)+AK(I+NUM*(K-2))*DR(I+NUM*(K-2)))/RP/RP/2.
40 AQTEN=8M(MN) ** (OTEN/QTBL)
10 CM=CMET(MM)/VT(MM)
   VB=VE3 (MM)/VT(MM)
 9 GO TO (43,44),1SP
43 AP=.75*(R(MN)+R(I+NUM*N))**2/(3.*DR(MM)*R(MN)**2+DR(MM)**3)
   AM = .75 * (R(MN) + R(1+NUM*(N-2))) * *2/(3.*DR(MM) * R(4N) * *2+DR(MM) * *3)
   60 TO 45
44 AP= (1.+R(I+NUM+N)/R(MN))/2./DR(MM)
   AM = (1. + R(I + NUM + (N - 2)) / R(MN)) / 2. / DR(MM)
45 GO TO (27,28), ISS
27 RC=1.
   GO TO 51
28 RC=RHO(MM) \star C(MM)
51 A(MN)=(-RHOB*CP*VB*AGTEN-AA*AM-BB*AP)/PC
   B(MN) = AA*AM/RC
   D(MN)=BB*AP/RC
   E(MN)=(CM*BM(MN)+RHOB*CB*VB*AQTEN*TAR(I))/RC
   GO TO (80,18), IGO
18 K=K+1
   IGO=1
BO CONTINUE
   DUV=DQ(I)*2.
   GO TO (33,34), ICEN
33 AKBOE=AK(I)
```

```
GO TO (49,50), ISP
34 AKBOB= (AK(I) *DR(I) +AK(I +NUM) *DR(I+NUM))/(DR(I)+DR(I *NUM))
   GO TO (49,50),1SF
49 ASF=6.
   50 TO 35
50 ASP#4.
35 GO TO (25,26),188
25 RC=1.
   60 TO 52
26 PC=RHO(1)*C(1)
52 A(I)=(-ASP*AKBOB/DUN/DUN-RHOB*CB*VEB(I)/VT(I)*FM(I)**(GTEN/QTEL))/
  *RC
   B(I)=0.0
   D(I) = A S P * A K B OB / RC / DUN / DUN
   E(I)=(CMET(I)/VT(I)*FM(T)+RHOD*CH*VEB(I)/VT(I)*TAR(I)*6M(I)**(QTEN
  */QTBL))/RC
   ISP=1
90 CONTINUE
   RETURY
   END
   SURROUTINE EBAL (KATT)
   IMPLICIT REAL+8(A-H-0-Z)
   DIMENSION IMIN(10), IMAX(10)
   DIMENSION BC(1),D4(1),R(1),V(1),VT(1),AC(1),RHO(1),C(1),AK(1),
  $PM(1),CMET(1),VFP(1),TAR(1),ID(1),FNGR(1),EBT(1),T(1),TT(1),QB(1),
  $8 FB (1)
   COMMON/ENV/BC/LDR/DR/LR/R/LV/V/LVT/VT/LAC/AC/LRHO/RHO/LC/C/LAK/AK
   COMMON/LEM/BM/LCMET/CMET/LVF6/VEB/LTAR/TAR/LBLD/RHOP/C9/LT/T
   COMMON/LID/NUM, ISPHER, NUMS, N, NOP, ID/LENGR/ENGR/LEBT/EBT/LBFB/BF3
   COMMON/LQTEN/QTEN/LQTBL/QTBL/LQB/QP
17 SUM= 2.
   STORED.
   BEC=O.
   GEN=0.
   I = 1
   CALL DEFRG(K, IMIN, IMAX, LMAT, NODE)
   NA=K+VUM*NODE
   DO 30 J=1,NODE
   IF(J.3T.IMAX(1)) I=I+1
   MN = K + VUM * (J - 1)
   M=K+NUM+(I-1)
   AQTEN=BM(MN) ** (QTEN/QTBL)
10 CM=CMET(M)+V(MN)/VT(M)
   VB = VE3(M) \times V(MN)/VT(M)
16 STUR=STOR+RHO(4) *C(4) *V(MN) *(T(MN)-TT(MN))/3C(K+NUM *4)
   REC = REC + RHO(M) * C(M) * V(MN) * (T(MN) - 273.16)
   ENGR (< + NUM * 6 )=BEC
   ENGR (C) = STOR
15 SUM = SUM + PHOB * CB * VB * (T(MN) = TAR(K)) * AOTEN
   FNGR (< + NUM) = SUM
3 GEN=GEN+CM+FM(MN)
   ENGR(K+NUM+2)=GEN
   OUTA = 9 C(K) * (T(NA) + BC(K + NUM * 2))
   ENGR(<+NUM*3)=OUTA
```

```
OUTB=3C(K+NUM)+(T(NA)++4-BC(K+NUM+5)++4)
    ENGR (C+NUM+4) mOUYB
   OUTC#3C(X+HUM#5)/T(MA)
   ENGR (< +NUM * 5 ) # OUT C
   OUT=OUTA+OUTB+OUTC
18 GO TO (13,14),N
13 EBT(K) = OUT-GEN+SUM
   RETURY
14 EBT(K) = OUT-GEN+SUM+STOR
20 RETURY
    END
    SUPROUTINE BLTPRG
   IMPLICIT REAL+8 (A-H, 0-2)
   DIMENSION IMIN(10) JIMAX (10) JOELT (126)
   DIMENSION V(1), VT(1), BM(1), BFB(1)
   DIMENSION VEB(1), TAR(1), T(1), ID(1), IVEN(1), UA(1), VELB(1)
   COMMON / L V/V / LVT / VT / L R FB / PFB
   COMMON/LVEB/VEB/LTAP/TAR/LBLD/RHOB/CB/TARP/LT/T/LBM/BM
   COMMON/LID/NUM:ISPHER,NUMS,ISS,NOP,ID/LTVEN/TVEC/LUA/UA/UA/LVELO/VFLB
   COMMON/LTIME/COLETIME/FWRITE
   COMMON/LATEN/ATEN
   COMMON/LQTBL/QTBL
   LOGICAL ASYFL
   SUMB=0.
   SUMC =0.
   DC 5 I=1,NOP
   IF(ASYFL(I)) GO TO 5
   SUMA=0.
   VEL9(I) = 0.
   CALL DEFRG(I, IMIN, IMAX, LMAT, NODE)
   1 = 1
   DO 15 K=1.NO DE
   (1-L) *MUM *I = TAM
   KN=I+VUM*(K-1)
   ARTEN=BM(KN) ** (RTEN/RTBL)
 4 VB=VE3 ("AT) * V(KN) /VT(MAT)
10 VELB(I)=VELB(I)+VB*AGTEN
25 SUMA=SUMA+T(KN) *VB*AOTEN
   1F(K.EQ.IMAX(J)) J=J+1
15 CONTINUE
   TVEN (I) = SUMA /VELB (I)
 5 CONTINUE
   DO 45 1=1,NOP
   IF(ASYFL(I)) GO TO 45
35 PELT(I) = TARP-TAR(I)
   TVEN(I) = TVEN(I) +DELT(I)
40 SUM C=SUM C+VELB(I) * TVEN(I)
   SUMB = SUMB+VELB(I)
45 CONTINUE
   CO=SUMB-CO
   TARP=SUMC/SUMB
   00 50 I=1.NOP
   IF(ASYFL(I)) GO TO 50
```

55 ALAM=UA(I)/VELB(I)/RHUB/CB

```
TAR(I) = (TARP +ALAM * TV FN (I))/(ALAM+1.)
59 CONTINUE
420 RETURN
    END
    SUBROUTINE STEPRO(IT,DD)
    IMPLICIT REAL+8 (A-H+0-Z)
    DIMENSION XAT(21)
    DIMENSION DD(1) - TAR(1) - EST(1) - ENGP(1) - ANAME(1)
    DIMENSION A(1),8(1),0(1),E(1),8((1),1(1),ID(1),TT(1),AT(1)
    COMMON/LTAR/TAR/LEBT/EBT/LENGR/ENGR
    COMMON / FNV/BC/COAYA/COB/B/COD/D/COE/E/LT/T
    COMMON/LTIME/COPETIME/EWPITE/LNAME/ANAME/LID/NUM/ISPHER/NUMS/ISS/N
   SOPAID
    COMMON/LSTOP/STOPIT
    IA=?
    LOGICAL ASYFL
    I 1 = 1
    IL=O
    CALL SCONT
53 17=2
    00 90 K=1.NOP
    YBC=BC(1+NUM*6)
    IF (ASYFL(K)) GO TO 37
    DD(K)=TAR(K)
   CALL DEFSRG(K,LMAT, NODE, IZ)
   NN = K + V UM + NO DE
   NU=VODE+1
12 DO 37 I=1.NU
   N=K+NUM*(J-1)
   XAT(I) = T(N)
   60 TO (31,30), IT
31 \text{ TT(V)} = \text{T(N)}
30 CONTINUE
52 IF(ISS_NF.1) GO TO 20
   T(K) = D(K)/A(K) + T(K+NUM) - E(K)/A(K)
   TT(K)=T(K)
   DO S I = S.NODE
   N=K+NUM*(I-1)
   T(N) = -E(N)/A(N) * T(K+NUM*(I-?)) * D(N)/A(N) * T(K+NUM*I) * -E(N)/A(N)
 6 \text{ (T(N)=T(N))}
   T(NN)=BC(K+NUM+と)*(BC(K)*(T(NN)-BC(K+NUM*2))*BC(K+NUM)*(T(NN)**4-
  $BC(K+NUM+3)++4)+RC(K+NUM+5)/T(NN))+T(K+VUM+(NODE-1))
   GO TO 41
20 T(K)=(-D(K)*EC(K*NUM*4)*T(K+NUM)-RC(K+NUM*4)*F(K)-TT(K))/(F(K)*BC(
  SK+NUM+4)-1.)
   DO 47 I=2,NODE
   N=K+NUM*(I-1)
40 T(N)=(-BC(K+NUM+4)+(E(N)+T(K+NUM+(I-2))+D(%)+T(K+NUM+I))+BC(K+NUM+
  $4) * E(N) - TT(N)) / (PC(K + NUM * 4) * A(N)-1.)
   T(NN)=-8C(K+NUM*8)*(8C(K)*(8C(K+NUM*2)-TT(NN))+8C(K+NUM)*(8C(K+NUM
  $*3)**4-TT(NN)**4)-BC(K+NUM*5)/TT(NN))+T(K+NUM*(NODE-1))
41 00 51 J=1,NU
   " (1-L) + MUN+X="
   TEST=DARS(XAT(J)-T(N))
```

```
(N)T = (L)TAX
    IF(TEST.GE.XBC) GO TO 52
51 CONTINUE
90 CONTINUE
    I T = 2
    IL=IL+1
35 CALL BLTPRG
    IAD=1
    ENERGY = 0.
    EMAT=0.
   DO 73 I=1.NOP
   IF(ASYFL(I)) GO TO 73
   ETEST=DABS(DD(I)-TAR(I))
   DD(I)=TAR(I)
   IF(ETEST.GE.RC(I+NUM*6)) IAD=2
   CALL EBAL (I,TT)
   ENERGY = ENERGY+EBT(1)
   EMAT=EMAT+ENGR(I+NUM+2)
73 CONTINUE
   CALL SCONT
   CALL COEFRG
   GO TO (74,53), IAD
74 IF(BC(1+NUM * 6). LE. STOPIT) 60 TO 77
   DO 75 I=1.NOP
75 BC(I+VUM+6) = EC(I+NUM+6)/10.
   60 TO 53
77 EMAT=.05*EMAT
   IF(DABS(ENERGY).LT.EMAT) GO TO 70
78 WRITE(3,79)
79 FORMAT(/1x, * CAUTION THE ENERGY BALANCE HAS NOT MET CRITERIA*)
70 IF(ISS.NE.1) GO TO 95
   WRITE(3,72) IL
   DO 1) K=1,NOP
   IF(ASYFL(K)) GO TO 10
   CALL DEFSRG (K, LMAT, NODE, IZ)
   NU=NODE+1
   DO 11 1=1.NU
   N=K+NUM*(I-1)
11 TT(N)=T(N)-273.16
   WRITE(3,71)ANAME(K),ANAME(K+NUM),ANAME(K+NUM*2),(TT(K+NUM*(1-1)),I
  S=1,4U)
14 CONTINUE
71 FORMAT (1x,3A4,1x,2(T15,11F10,5/))
72 FORMAT C/1X. STEADY STATE CONDITIONS HAVE BEEN REACHED ...... THE
  SNUMBER OF ITERATIONS EXECUTED ==>',110//1x' THE STEADY STATE TEMPE
  SRATURE DISTRIBUTION FOLLOWS, 1/1x, STARTING FROM THE CENTER NODE to
  * THE SURFACE NODE ==>'/)
   188=2
95 RETURY
   END
   SUPROUTINE GEOMRG (GEO, NUT, NOR)
   IMPLICIT REAL+8(A-H,0-Z)
   (0 1) d.(ut) xAMI (CT) MIMI. (BCM.TUM)CBD MOISHBMID
   DIMENSION DR (1),R(1),V(1),VT (1),AC (1),ID(1), TV(1),ANAME (1)
```

```
COMMON/LDR/DR/LR/R/LV/V/LVT/VT/LAC/AC/LJD/NUM,ISPHER,NUMS,ISS,NOP,
      $1D
        COMMON/LIV/YV/LICT/ICTV/TOTA/LNAME/ANAME
       LOGICAL ASYFL
        DATA P1/3.141592/
        TCTV=).
        C. C=ATOT
       WRITE(3,68)
68 FORMAT ("1", T 26, "SKIN
                                                                SUB C UT
                                                                                      DIA
                                                                                                         DIA
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                                                             SPHERE" /T7, "ELEMENT
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     SDIA
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     20)
                  SEGMENT'/T18, (CM)
                                                                      THICK
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                                                                                                                                                    MATL
                                                                                                              ANGLE 1/ T27 (CM)
                                                  MATL
                                                                      MATL
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              MATL
                                MATL
                                                                  5-6
                                                                                                       7-8
     $ (CM)
                             3-4
                                               4-5
                                                                                    6 - 7
                                                                                                                         8-9
                             (RAD) 1)
     $ (CM)
       NUMS=VOP
       DO 31 I=1,NOP
       ISEG=2
       IF(ASYFL(I)) GO TO 30
       TV(I)=0.0
       READ(1,61)(GFO(I,J),J=1,12)
51 FORMAT(F10.6,1X,F6.5,1X,F6.4,7F5.3,F8.4,F10.7)
       62 FORMAT (1X,3A4,1X,F17.6,1X,F6.5,1X,F8.5,1X,8F8.4,1X,F19.7)
       CALL DEFRG(I, TMIN, IMAX, LMAT, NO DE)
       NU=NODE+1
       DA=GED(1,1)/PI
       D(3) = 0 A - GEO(1.3)
       K=LMAT+1
       00 15 J = 4,10
       D(J)=D(J-1)*GEO(I-J)
       N=K-J+1
       IF (J.EQ.K) GO TO 17
15 DR(I+NUM*(N-1))=(D(J-1)-D(J))/4./DFLOAT(IMAX(N)-IMIN(N)+1)
17 IE=2*(IMAX(N)-IMIN(N)+1)-1
       DR(I+NUM*(N-1)) = (D(J-1)-P(J))/2./DFLOAT(IE)
       DR(I+YUM+(LMAT-2))=(GEO(I,3)-GEO(I,2))/4./DF10AT(IMAX(LMAT-1)-IMIN
     3 (LMAT-1)+1)
       DR(T+VUM+(LMAT-1))=GEO(I_2)/4./DFLOAT(TMAX(LMAT)-IMIN(LMAT)+1)
       SFG=GEO(1,12)/2./PI
       K = 1
       VT(I+VUM*(K-1))=0.
       C=(I)q
       R(I+NJM+(NU-1))=DA/2.
       DO 10 N=2,NOTE
       IF(N.ST.IMAX(K)) GO TO 11
12 R(I+NUM*(N-1))=R(I+NUM*(N-2))+DD
       IF(I.GT.ISPHER) GO TO 33
      ISEG=1
       SEG=GEO(1,12)*PI*(1.-DCOS(GEO(1,11)*PI))
      V(I+NUM*(N-1)) = ((R(I+NUM*(N-1))+DR(I+NUM*(K-1)))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R(I+NUM*(N-1)))**3-(R
    $)-DR(I+NUM+(K-1)))++3)+SEG/3.
      60 TO 34
```

```
33 V(I+NUM+(N-1))=PI+((P(I+NUM+(N-1))+DR(I+NUM+(K-1)))++2~(R(I+NUM+(N
  5-1))-3R(I+NUM*(K-1)))**2)*GEO(I,11)*SEG
34 VT(I+VUM*(K-1))=VT(I+NUM*(K-1))+V(I+NUM*(N-1))
   TV(I) = TV(I) + V(I + NUM + (N + 1))
   CO TO 10
11 K=K+1
   DD=DR(I+NUM*(K-1))+DR(I+NUM*(K-2))
   VT(I+VUM*(K-1))=().
   60 TO 12
1 CONTINUE
   SO TO (35,36), I SEG
36 V(I)=PI *DR(I)*DR(I)*GEO(I,11)*SEG
   AC(I)=GEO(I,1)*GEO(I,11)*SEG
   60 TO 37
35 V(I)=DR(I)*DR(I)*DR(I)*SEG/3.
   AC(I)=R(I+NUM*NODE)**2*SEG
37 VT(1)=VT(1)+V(1)
   TV(I) = TV(I) + V(I)
   GO TO 31
3 > YUMS=YUMS-1
31 CONTINUE
   DO 70 N=1,NOP
   IF(ASYFL(N)) GO TO 70
   CALL DEFSRG(N,LMAT,NODE,2)
   NII=NODE+1
   WRITE(3,63)N,ANAME(N),ANAME(N+NUM),ANAME(N+NUM*2),AC(N),TV(N)
   WRITE(3,64)(VT(N+NUM*(K-1)),K=1,LMAT)
   WRITE(3,65)(DR(N+NUM*(K-1)),K=1,LMAT)
   WRITE(3,A6)(P(N+NUM+(K-1)),K=1,NU)
   WPITF(3,67)(V(N+NUM*(K+1)),K=1,NODE)
   TOTV=TOTV+TV(N)
   TOTA=TOTA+AC(N)
70 CONTINUE
63 FORMAT(//T10, "DATA FOR BODY ELEMENT", 16, 2×, 3A4/T10, "SUPFACE AREA (
  #SU.CM) 1,2X,F10.3,T65, TOTAL VOLUME (CUBIC-CM) 1,4X,F10.3/)
54 FORMAT (3X/*VT (CUBIC-GM)*/2X/(T20/10F10.3))
55 FORMAT (3Y, 'DR (CM) ', 2X, (T20, 10F10.3))
66 FORMAT (1x, RADIUS (CM) 1,2x, (T20,10F10.4))
57 FORMAT (1x, VOLUME (CUBIC-CM) 1,2x, (T20, 1)F10.3))
   TOTA=TOTA/10000.001
   WRITE(3,69) TOTV, TOTA
69 FORMAT (///1x, TOTAL BODY VOLUME==> *, F10.3, * (CUBIC CM) *, T60, *TOTAL
  $ BODY AREA ==> ", F10.3," (SQU 'RE METERS) "/)
   RETURN
   END
   SUBROUTINE OFFRG(LSEC, IMIN, IMAX, LMAT, NODE)
   IMPLICIT REAL+S (A-H,O-Z)
   DIMENSION IMIN(10), IMAX(1)), ID(126)
   COMMON/LID/NUM, ISPHER, HUMS, ISS, NOP, ID
   17=1
   ENTRY DEFSEG (LSEC, LMAT, NODE, 17)
   NO=ID(LSEC)
   NCDF=J
   DC 15 I=1,10
```

```
NE =NO/10
      ICC=NO-NE+10
      NO=NE
      IF (ICC.FO.0) 50 TO 18
      IMIN(1) = ICC
   15 NODE=WODE+ICC
      1=1+1
   18 J=I-1
      LMAT =J
      60 TO (20,30),12
  3 RETURY
  20 ISTOR=IMIN(1)
      IMIN(9)=1
     IMAX(1)=ISTOR
     DO 16 K=2,J
     ISTOR=IMIN(K)
     IMIN(<) = IMAX(K-1)+1
  16 JMAX (<) = IMIN (K) + ISTOR-1
     RFTURV
     END
     LOGICAL FUNCTION ASYFL(1)
     COMMON/LIA/IA(1)
     IF(I.VE.IA(I)) GO TO 5
     ASYFL= . FALSE.
     RFTURY
  5 IF(1.37.1A(1)) GO TO 6
    ASYFL= . FALSE .
    RETURY
  6 ASYFL= . TRUE .
    RETURY
    END
    SUBROUTINE MUSCLE (I.J.K.L.M)
    DIMENSION IMU(1)
    COMMONATIMANIMA
    J = IMU(I) - (IMU(I)/10) * 10
    K=(IMU(I)-(IMU(I)/100)*100)/10-1
    L=(IMU(I)~(IMU(I)/1000)*1000)/100-1
    M=IMU(I)/1000-1
    RETURY
    END
    SUPROUTINE TERPHG (TABLE, COTAB, GIVE, WANT, INDEX)
   IMPLICIT REAL+8 (A-H,0-Z)
   DIMENSION TABLE (INDEX). COTA 3 (INDEX)
   IF(GIVE.LT.TABLE(1)) GO TO 75
   IF (GIVE.GT.TABLE (INDEX)) GO TO 35
   IF (INDEX.GT. 10) GO TO 40
   L=1
   WANT=COTAB(L)
   PROD=1.
   IND=INDEX-1
   60 TO 18
40 00 13 ILA=1, INDEX
   LAFILA
   IFCTABLE(LA) .GE.GIVE) GO TO 45
```

```
13 CONTINUE
 45 IF (TABLE (LA) .EQ .GIVE) GO TO 63
    IF (LA.LE.10) IND=10
    IF(LA.ST.10) IND=LA
    L=1ND-9
    WART = COTAR(L)
    PROD=1.
 18 DO 19 I=L.IND
    PROD=PROD*(GIVE-TABLE(I))
    DIVDIF=0.
    IA= I +1
    DO 30 K=L.IA
    DFR30=1.
    DO 25 J=L,IA
    IF(J.EQ.K) GO TO 25
    DPROD=DPROD*(TAPLE(K)-TABLE(J))
 25 CONTINUE
 3 DIVDIF = DIVDIF+COTAE(K)/DPROD
 1 " WANT=WANT+PROD*DIVDIF
    RETURN
 15 WANT=COTAB(1)
    WRITE(3,50) GIVE
    RETURY
 35 WANT=COTAB(INDEY)
    WRITE(3,50) GIVE
 50 FORMAT (1x, THE VALUE BEING INTERPOLATED (", £16,8,") EXCEEDS TABLE
   SRANGE, STANDARD FIXUP TAKEN )
    RETURN
 61 WANT =COTAB(LA)
    RETURN
```

END

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APPENDIX C

THE MONTGOMERY MODEL (MODIFIED)

The Montgomery model was implemented as a FORTRAN program. It has run on both an IBM 370/168 and an IBM 3033. On the IBM 3033 a ninety-minute simulation requires approximately twenty-eight CPU seconds. The following table lists the major variables of this model. Following the listing of the model. Its input data and a typical output, there are instructions for using it.

MAJOR VARIABLES OF THE MONTGOMERY MODEL

Symbol .	Definition	Units
A(N)	Area of the mid-plane between compartment N and N+1	m ²
AL(I)	Wet suit half-thickness covering segment I	м
AT	Adipose tissue weight	KG
BFB(N)	Basal blood flow to compartment N	L/HR
BFWT	Body fat weight	CIM
RM	Diver's basal matabolic rate	KCAL M -HR
C(N)	Thermal capacitance of compartment N	KCAL/OC
CM(N)	Radius to the center of mass of compartment N	М
10(1)	Outside diameter of Segment I	М
HAR	Height of arms segment	M
r if	Height of feet segment	M
нна	Height of hands segment	M
HL	Height of legs segment	M
HSS(I)	Thermal conductance between skin and wet suit of segment I	KCAL/HR-OC
нт	Diver's height	CM
ETR	Height of trunk segment	M
K(L)	Thermal conductivity values of different materials	KCAL M-HR- ^O C
LBWT	Lean body weight	GM
NAT	Non adipose tissue weight	KG

Symbol	Definition	Units
PBF	Diver's percent body fat	N.D.
QB(N)	Basal metabolism of compartment N	KCAL/HR
QC	Total basal metabolism of body core excluding head and trunk core	KCAL/HR
QM	Total basal metabolism of muscle	KCAL/HR
QSF	Total basal metabolism of skin plus fat	KCAL/HR
R(N)	Radius of compartment N	М
rmp(n)	Radius to the mid-plane between com- partment N and N+1	м
s(I)	Outside surface area of segment I	WS
SA	Total body surface area	cm ²
SG	Specific gravity of diver	N.D.
SWT(N)	Body compartment (N) weight	KG
TC(N)	Thermal conductance between compartment N and N+1	KCAL/HR-°C
THWS(I)	Wet suit thickness covering segment I	М
VAR	Volume of arms segment	м3
VF	Volume f feet segment	_M 3
VHA	Volume of hands segment	_M 3
VL	Volume of legs segment	м ³
VTR	Volume of trunk segment	_M 3
WC	Weight of body core	KG
WM	Total weight of mascle	KC
WSCP	Wet suit specific heat	<u>kcal</u> kg- ^o c
WSD	Wet suit density	кс/м ³
WSF	Total weight of skin plus fat	KG
WT	Diver's weight	GM
x(n)	Heat transfer length between compartment N and N+1	M

Symbol	Definition	Units
COND3	Total body heat loss to smbient water	KCAL M ² -HR
DEN	Density of ambient water	kg/m³
DILAT	Total efferent skin vasodilation command	L. HR
יתם	Integration step	HR
E(N)	Total evaporative heat loss from N	KCAL HR
EB(N)	Basal evaporative heat loss from N	KCAL/HR
E E	Total respiratory heat loss	KCAL HR
ERROR(N)	Output from thermoreceptors in compartment N	°c
EV	Total evaporative heat loss	KCAL M ² -HR
F(N)	Rate of change of temperature in N	°C/HR
H(I)	Total environmental heat transfer coefficient for Segment I	KCAL M ² -HR-OC
HF(n)	Rate of heat flow into or from N	KCAL HR
HFLOW	Total rate of heat flow to or from the body	KCAL M²-HR
HO(I)	Water-wet suit surface heat transfer coefficient for Segment I	M ² -HR-°C
HP	Total metabolic heat production	KCAL M ² -HR
HTBAL	Total body heat balance	KCAL M ² -HR

Symbol	Definition	Units
R200UT	Total water loss rate in expiratory gas	GM HR
INT	Interval between outputs	MIN.
ITIME	Elapsed time	MIN.
JTIKE	Elapsed time	MIN.
М	Metabolic heat production = HP (Print out label)	<u>KCAL</u> M ² -∴IP.
MU	Dynamic viscosity of water	<u>КС</u> Н R-М
P(I)	Water vapor pressure table from 0-50°C	mm Hg
PAIR	Vapor pressure of environmental air	mm Hg
PIN	Saturated water vapor pressure of inspired air	mm Hg
POUT	Saturated water vapor pressure of expired air	num Hg
PR	Prandtl Number of environmental water	N.D.
Q(N)	Total metabolic heat production in N	KCAL HR
RATE(N)	Dynamic sensitivity of thermo- receptors in N	HUR
RE(I)	Reynolds' Number of body Segment I	N.D.
RHEAT	Respiratory heat generation rate	CAL MIN
RHL	Respiratory heat loss rate	KCAL HR
S	Nate of body heat storage = HFLOW (Print out Usbel)	KCAL M ² -HR
s(1)	Surface area of Segment I	m ²

Symbol .	Definition	Units
SA	Total skin surface area	M ²
SBF	Total skin blood flow	L MIN
ekinc(I)	Fraction of vasoconstriction command applicable to skin of Segment I	N.D.
skinr(i)	Fraction of all skin receptors in Segment I	N.D.
skins(I)	Fraction of sweating command applicable to skin of Segment I	N.D.
skinv(i)	Fraction of vasodilation command applicable to skin of Segment I	N.D.
STRIC	Total efferent skin vasoconstriction command	N.D.
sv	Specific volume of water	м ³ /кс
SWEAT	Total efferent sweat command	KCAL HR
T(N)	Temperature of N	°c
TB	Mean weighted body temperature	°C
TD(N)	Conductive heat transfer between N and N+1	KCAL HR
TE	Absolute temperature of expiratory gas	°K
TH	Temperature of the head core, representing the hypothalamic temperature = T(1) (Print out label)	°c
TI	Absolute temperature of inspiratory gas	°ĸ
TIME	Elapsed time	MIN.
TM	Temperature of muscle compartment in the leg = T(47) (Print out label)	°c

Symbol	Definition	Units
TO	Temperature of the central blood compartment, representing esophageal temperature T(61)	- °c
	(Print out label)	C
TR	Trunk core temperature representing rectal temperature = T(ll) (Print out.label)	°c
TS	Mean weighted skin temperature	°c
TSET(N)	"Set point" or reference point for receptors in compartment N	°c
U(I)	Overall body Segment I wet suit- water heat transfer coefficient	KCAL M ² -HR-°C
VE	Respiratory gas expiration rate	L HR
VEMIN	Respiratory gas expiratory minute volume	L MIN
WARM(N)	Output from warm receptors in N	°c
WARMS	Integrated output from skin warm receptors	°c
WORKM(I)	Fraction of total work done by muscles in Segment I	N.D.
BULL	Factor determining temperature sensitivity of sweat gland response	1/°c
CCHIL	Shivering from head core	KCAL/HR-°C
CCON	Vasoconstriction from head core	1/°C
CDIL	Vasodilation from head core	L/HR-OC
CSW	Sweating from head core	KCAL/HR-OC
SCHIL	Shivering from skin	KCAL/HR-OC
SCON	Vasoconstriction from skin	ı/°c
SDIL	Vasodilation from skin	L/HR-°C KCAL/HR-°C
SSW	Sweating from skin	KCAL/HR-°C

Symbol .	<u>Definition</u>	Units
PCHIL	Shivering from skin and head core	1/°c2
PCON	Vasoconstriction from skin and head core	1/°c ²
PDIL	Vasodilation from skin and head core	L/HR-°C2
PSW	Sweating from skin and head core	KCAL/HR-OC

```
COMMON / PASS/AL (6) > DO (6) > HSS (6) > S(6) > TC (60) > Q P(60) > B FB (60)
     */C(67)/CS(1)/CH(1)/RK(6)
      CALL SIZE
      CALL WEIMAN
       STOP
      END
      SUBROUTINE SIZE
C THIS PROGRAM IS FOR A FULL WET SUIT HEAD OUT
       REAL VATAK
       DIMENSION K(3)
      DIMENSION X (60) , RMP (60) , A(60)
      DIMENSION THWS(0), R(56), CM(66), H(6)
      DIMENSION SWI(61)
      COMMON/PASS/AL(6),DO(6),HSS(6),S(6),TC(60),QB(60),BFB(60)
     *,C(67),CS(1),CB(1),RK(6)
      DATA #SCP/0.25/, WSD/241.0/, SG/0.0/
      DATA THWS/0.006312/0.006312/0.006312/0.006312/0.006312/0.006312/
      DATA K/0.3193,0.1419,0.046/
      DATA RK/0.03711/0.03711/0.03711/0.03711/0.03711/0.03711/0.03711/
      READ(1,3)HT, WT
3
      FGRMAT (2F1U.5)
      REAL (1,7) PM
7
      FORMAT (F10.5)
      1F($G.GT.0.0)G0 TO 5
      SG=0.8*((HT**0.242)/(WT**0.1))+0.162
5
      PBF=(5.548/SG)-5.044
      BFWT=PBF *WT
      LBWT=AT-BFWT
      SA=71.84*((WT/1000)**0.425)*(HT**0,725)
      $(1)=D.07*$A*0.0001
      1000.3 *A2 *S06.0=0001
      S(3) = 3.1341 * SA * 0.0001
      S(4)=J. 75*SA +C. 5801
      $(5)=0.3174 * SA * 0.0001
      $(6)=J.0686*SA*0.0001
      AT=REAT +U.QU1
      NAT=L3WT*U.OU1
      DO 2 I=1.4
      SWT(I)=(0.0164/(0.85*4.0))*NAT+(0.024/(0.85*4.0))*NAT
2
      DC 4 J=5.8
      SWT(I) = (0.005/(0.85*4.0))*N'T
      SWT(9) = (0,005/0.15) * AT
      SWT(10) = (0.0036/7.85) *NAT
      DO 6 I = 11,14
      SWT(1) = (0.038/(0.85*4.02)*NAT+(0.159/(0.85*4.0))*NAT
6
      0081=15.18
٩
      SWT(1) = (7.24/9/(7.85+4.0)) + NAT
      SWT (19) = (U.095/U.15) * 4T
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```
TAR*(23) = (0.0181/0.85) *NAT
       00 10 I=21,24
10
       SMT(I) = (0.02028(J.85*4.0))*NAT+(0.01/(0.85*4.0))*NAT
       85.25ml St 0d
12
       SWT(I) = (0.0453/(0.85*4.0))*VAT
       SWY(29) = (0.013/0.15) *AT
       TAN*(35)=(0.0065/0.85)*NAT
       DO 14 1=31,34
14
       SUT(1) = (0.0031/(0.85*4.9))*NAT+(0.0004/(0.85*4.0))*NAT
       DO 16 I = 35,38
16
       SWT(I) = (0.001/(0.85*4.0))*NAT
       SWT(39) = (0.002/0.15) * AT
       SWT(43) = (0.0025/0.85) *NAT
       DO 150 I=41,44
       SWT(1) = (0.0673/(0.85*4.0))*NAT+(0.0258/(0.85*4.0))*NAT
180
       DO 200 I=45,48
200
       SuT(1) = (0.1368/(0.85*4.0))*VAT
       SWT(49) = (0.032/0.15) *AT
       SWT(50) = (0.0161/0.85) * NAT
       DO 22 I=51,54
55
       SWT(I) = (0.005/(0.85*4.0))*NAT+(0.0008/(0.85*4.0))*VAT
       DO 24 1=55,58
24
       SWT(1) = (0.001/(0.85*4.0))*NAT
       SWT(59) = (0.003/0.15) *AT
       SWT(60) = (0.0032/0.85) * NAT
       SWT(61)=2.50
       00 26 I=1.4
       C(1)=(0.5*(C.C164/0.85)*NAT+0.9*(0.024/0.85)*NAT)/4.0
26
       DO 28 I=5.8
       (I) #3.9*SWT(I)
85
       C(9)≈0.6*SWT(9)
       C(10) = 0.9 + SWT(10)
       00 300 1=11.14
       c(1) = (0.5*(0.038/0.85)*NAT+0.9*(0.159/0.85)*NAT)/4.0
300
       00 323 I=15,18
320
      C(I) = 0.9 * SWT(I)
       C(19)=0.6*SWT(19)
       0(20) = 0.9 \times SWT(20)
       DO 340 1=21,24
340
      C(1) = (0.5*(0.0202/0.85)*NAT+0.9*(0.01/0.85)*NAT)/4.0
      DO 36 I=25,28
36
      C(1) = D . 9 = SWT(1)
      C(29)=0.6*SWT(29)
      C(30) = 0.9 \times SWT(30)
      DO 38 1=31,34
38
      C(1) = C0.5 \times (0.0031/0.85) \times AT + 0.9 \times (0.0004/0.85) \times AT)/4.0
      DO 4.1 X = 35, 38
40
      (I) Tw2*? [=(I))
      C(39) = 0.6*SWT(39)
      €(47)=0.9×SWT(40)
      DO 42 1=41,44
42
      C(1) = (0.5*(0.3673/0.85)*(AT+0.9*(0.0258/0.85)*(AT)/4.0
      DO 44 1=45,48
      C(1)=0.9*SWT(1)
44
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C(49)=0.6*SWT(49)
       C(50)=0.9*SWT(51)
       00 46 I=51,54
       C(I)=(O.5*(O.005/O.85)*NAT+0.9*(O.0008/0.85)*NAT)/4.0
46
       00 48 I=55,58
48
       (1) = 3.9 \times SWT(1)
       C(59) = 0.6 * SWT(59)
       C(60)=0.9*5WT(60)
       C(61)=0.9*SWT(61)
       DO 50 I=11,14
50
       C(T) = C(T) + C(61)/4.7
       CS(1)=C(19)+C(29)+C(39)+C(49)+C(50)+C(69)
       CB(1)=0.0
       DO 52 I=1.61
       CB(1) = CB(1) + C(I)
52
       CONTINUE
       GSF=0.3*(0.05882*NAT+1.0*AT)
       QM=(0.18*(8M*SA/10000))-QSF
       QC=0.10*(BM * SA/10000)
       WSF=0.05882*NAT+1.0*A1
       ₩Y=0.5059*NAT
       WC=0.1765*NAT+0.2588*NAT
       DO 54 I=1.4
54
       QB(I)=SWT(I)*QC/WC+0.04*(BM*SA/10000)
       DO 56 I=5,8
56
       QB(I)=SWT(I) *GM/WM
       GP(9)=SWT(9) *QSF/WSF
       QB(10) = SWT(10) * USF/4SF
       DO 58 I=11.14
58
       QB(T) = SWT(T) *QC/WC+D.14*(BM*SA/10000)
       DO 67 1=15,18
60
       QR(I) = SWT(I) *QM/WM
       Qb(19) = SWT(19) \star RSF/WSF
      QB(2)) = SWT(20) + QSF/WSF
       DO 62 I=21,24
62
      GB(I)=SWT(I) *QC/WC
      DO 64 1=25,28
64
      QD(I) = SWT(I) *QM/AM
      DO 66 1=29,30
66
      QB(I)=SWT(1) *QSF/WSF
       00 68 1 = 31, 34
      QB(1)=SWT(1) +0C/WC
68
      00.70 I = 35.3a
70
      QB(I)=SWT(I)+QM/WM
      DO 72 1=39,41
72
      QB(I)=SWT(I)+QSI/WSF
      DO 74 1=41,44
74
      98(1) = SW7(1) *QC/9C
      00 76 1=45,45
76
      QH(I) = SHT(I) *QM/HM
      DO 78 1=49.50
78
      QB(I)=SWT(I) *GSF/WSF
      DO 87 I=51,54
8.0
      QB(1)=SWT(1) +0C/WC
```

```
DO 82 I=55.58
82
      QB(I) = SWT(I) + QM/WM
      DO 84 1=59,6
      QB(I)=SWT(I) *USF/WSF
84
       DO 86 I=1,60
      BFE(I) = 1.2 * QB(I)
86
      DO 88 I=1.4
      BFB(I)=11.25
88
      DO 97 I=11,14
90
      BFB(1) = 52.5
      BFB(10) = 5.34 * SVT(10)
      BFR(23)=1.56*SWT(20)
      BFB(30) = 1.35 * SWT(30)
      BFB(40) =5.58 +SWT(40)
      BFD(50) = 0.84 * SWT(50)
      BFB(60) = 2.34 * SWT(60)
      R(1)=(((0.003*SWT(1))/12.56)**0.333)
      DO 92 I=2,10
      R(I) = ((R(I-1)**3.0)*((0.003*SWT(I))/12.56))**0.333
92
      R(61) = R(10) + THWS(1)
      VTR=0.0
      DO 94 I=11,20
94
      VTR=VTR+SWT(I)*0.001
      HTR = (S(2)**2.0)/(12.56*VTR)
                          A IS PREVIOUSLY DIMENSIONED.****
C*****CHANGE A TO AA.
      10C.0=AA
      R(2()) = (2*VTR)/S(2)
      DO 96 1=1.9
      J = 21 - 1
      R(J-1) = ((R(J)**2) - ((AA*SWT(J))/(3.14*HTR)))**0.5
96
      CONTINUE
      R(62) = R(20) + THWS(2)
      VAR=U. ()
      DO 98 1=21.3.
98
      VAR=VAR+SUT(1) + 0.931
      HAR= (S (3) **2.0) / (12.56 * V/E)
      R(30) = (2.0 \pm VAH)/S(3)
      DO 100 1=1-9
      1=31-1
      R(J-1) = ((R(J)++2)-((AA+SWT(J))/(3.14+HAR)))++0.5
1 30
      CONTINUE
      R(63) = R(30) + THWS(3)
      VHA=0.0
      DO 102 T=31,40
      VHA=VHA+SWY(I) * 0.001
102
      HHA= (5(4)**2.0)/(12.56*VHA)
      R(40)=(2.0+VHA)/S(4)
      DO 1 14 1=1.9
      J = 41 - 1
      R(J-1) = ((R(J)**?)-((/A*SbT(J))/(3.14*HHA)))* *G.5
104
      CONTINUE
      R(64)=R(40)+THWS(4)
      VL=0.3
      DO 136 1=41,50
```

```
1 96
        VL=VL+ SWT(I) +0.001
        HL=(S(5)**2.()/(12.56*VL)
        R(50) = (2.0 * VL) / S(5)
        DO 178 I=1.9
        J = 51 - I
        R(J-1) = ((R(J)**2) - ((AA*SWT(J)))/(3.14*HL)))**U.5
 108
        CONTINUE
        R(65) = R(50) + THWS(5)
        VF=0.0
        DO 110 I=51,60
 110
       VF=VF+ SWT(I) *0.001
       HF=(S(6)**2.0)/(12.56*VF)
       R(60) = (2.0 * VF) / S(6)
       DO 112 I=1,9
       J=61-I
       R(J-1) = ((R(J)**2) - ((AA*SWT(J)))/(3.14*hF)))**(1.5)
 112
       CONTINUE
       R(66) = R(60) + THWS(6)
       CM(1)=0.7937*R(1)
       DO 114 I=2,1U
 114
       CM(I)=0.7937*((R(I-1)**3)+(R(I)**3))**0.333
       CM(61)=0.7937*((R(10)**3)+(R(61)**3))**0.333
       C^{M}(11) = 0.7071 * R(11)
       DO 116 I=12,20
 116
       CM(I)=0.7071*((R(I-1)**2)+(P(I)**2))**7.5
       CM(62) = 0.7071*((R(20)**2)*(R(62)**2))**0.5
       CM(21) = 0.7071 * R(21)
       DO 118 T=22,30
118
       CM(I) = 0.7071 * ((R(I-1)**2)*(R(I)**2))**0.5
       CM(63) = 0.7071*((R(30)**2)*(R(63)**2))**0.5
       CM(31) = 0.7071 * R(31)
       00 120 1=32,40
120
       CM(1)=0.7071*((P(I-1)**2)+(R(I)**2))**0.5
       Cil(64) = 2.7071*((R(4)) **?)+(R(64)**?))**n.5
       CM(41) = (.7071*R(41)
       DO 122 I=42,50
122
       CM(I) = 0.7071 * ((R(I-1)**2)* (%(I)**2))**0.5
       CM (65) = 0.7071*((R(5)) **2)+(R(65) **2)) **0.5
       CM(51) =0.7071*R(51)
       DO 124 I=52,60
124
       CM(I)=0.7071+((R(I-1)++2)+(R(I)++2))++6.5
       CM(66) = 0.7071*((K(67)**2)+(@(66)**2))**0.5
       DO 126 1=1.60
126
       X(I) = CM(I+1) - CM(I)
       DO 128 1=1.6
       X(10*L) = CM(1+67) - CM(10*1)
128
       DO 130 I=1.69
      RMP(I) = CM(I) + x(I)/2.0
130
      DO 132 I=1,16
132
      A(I)=12.56*(RMP(I)**2)
      00 134 1=11,20
134
      A(I)=6.28*RMF(I)*HTR
      00 136 I=21,39
136_
      A(I)=5.28*RMP(I)*HAR
```

```
DO 138 1=31,40
      A(I)=6.28*RMF(I)*HHA
138
      DO 140 1=41,50
140
      A(I)=5.28*RMP(I)*HL
      DO 142 1=51,60
      A(I)=6.28*RMP(I)*HF
142
      DO 144 I=1.00
      LC(I) = (K(1) * Y(1))/X(1)
144
      RMPA = CM(8) + ((R(9) - CM(8))/2.7)
      AA=12.56*(RMPA**?)
      XA = R(S) - CM(S)
      YA = (K(1) * AA)/XA
      RMPB=R(8)+((CM(9)-R(8))/2.0)
      AB=12.56*(RMP6**2)
      XE=CM(9)-R(8)
      YB = (K(2) + AB)/XB
      \Upsilon C(8) = (\Upsilon A * \Upsilon B)/(\Upsilon A + \Upsilon B)
      PO 145 I=1.6
      TC(10*1)=0.0
146
      DO 148 I=1.6
148
      TC(10*I-1)=(K(2)*A(10*I-1))/X(10*I-1)
      H(1)=0.0
      H(7)≈HTR
      H(3)=HAR
      H(4)=+HA
      H(5)=HL
      H (6)=H F
      DO 150 I=2.6
      RMPA1=CM(10*1-2)+((R(10*1-2)-CM(10*1-2))/2.0)
      AA1=6.28*RMPA1*H(1)
      XA1=R(10*I-2)-CM(10*I-2)
      YA1=(<(1)*AA1)/XA1
      RMPR %= R (10*1-2)+((C%(10*1-1)-R(1(*1-2))/2.0)
      A01=6.28*RMP81*H(1)
      YB1=C4(10+1-1)-R(10+1-2)
      AR1=(K(5)*WB1)\XU1
      T((10*1-2)=(Y/1*Y^{1})/(YA1*YP1)
150
      CONTINUE
      SMPR=CM(10)+((R(10)-CM(10))/2)
      SMPA=12.56*(SMPR**?)
      SX=R(10)-CM(10)
      SY = (K(2) + SMP /) / SX
      WSMPR=R(10)+((CM(61)-R(19))/2)
      WSMPA=12.56 * (WSMPR**2)
      WSX=CM(61)-R(10)
      WSY= (K (3) *WSMPA)/WSX
      HSS(1) = (SY+WSY)/(SY+WSY)
      DO 152 1=2.6
      SMPRI=CM(10*1)+((R(10*1)-CM(10*1)))/2)
      SMP#1=6 28+ SMPRI+H(I)
      SX1=R(1()*1)-CF(1()*1)
      SY1=(K(2)*SMPA1)/SY
      WSMPR1 = R(10 + 1) + ((CM(1 + 61) - R(1) + 1))/2.5)
      WSMPA1 = 6.26 + WSMPR1 + H(I)
```

```
WSX1=CM(I+60)-R(10*I)
       WSY1=(K(3)*WSMPA1)/WSX1
       HSS(I) = (SY1 + WSY1) / (SY1 + WSY1)
152
       CONTINUE
       DC 154 I=1.6
154
       AL(I) = R(I+6C) - CM(I+6C)
       DO 156 I=1.6
156
       DO(1) = 2.0 + R(1 + 60)
       S(1)=12.56*(R(61)**2)
       DO 158 1=2.6
158
       S(I)=6.28*R(I+e^{-})*H(I)
       C(62) = WSCP + WSD + 1.333 + 3.14 + ((R(61) + 3) - (R(10) + 3))
       0,5=1 Cot 0d
160
       C(1+61) = WSCP + WSD + ((R(1+60) + 2) - (R(10+1) + 2)) + H(1) + 3.14
       SAA=SA / 10000 -C
       VHEAD=0.0
       DO 162 I=1.10
       TOOLC*(I) TWZ + DABHV = DABHV
162
       WRITE(3,361) SG, PBF, SAA
361
       FORMAT (1x, BODY S.G.= ", F7.5, 2x, "% BODY FAT= ", F7.4,
      &2X2 SURFACE AREA= 1, F7.4, SQ M1)
       WRITE(3,37) AT, NAT
37
       FORMAT (/, 1x, 'ADIPOSE TISSUE WT. = ', F7.4, 1x, '< G', 2x,
      & 'NON-ADIPOSE TISSUE WT. = ', F7.4, 1X, 'KG')
       WRITE(3,30)R(10), VHEAD
30
       FORMAT (/,1X, "HEAD RADIUS=",F7.4,1X,"M",T30,"HEAD VOL.=",
      8 F7 . 4 . 1 X . 'M CUBE')
       WPITE(3,31) HTR, VTR
31
       FORMAT (1x, TRUNK LENGTH= 1, F7.4, 1x, TM1, T30, TPUNK VOL. = 1,
      8 F 7 . 4 . 1 X . 'M CUPE ')
       WRITE(3,32)HAR, VAR
32
       FORMAT (1X, ARM LENGTH= 1, F7.4, 1X, 14 1, T30, ARM VOL.= 1,
     $ F7.4, 1x, 'M CUBE')
       WRITE(3,33) HHA, VHA
33
       FORMAT (1X) 'HAND LENGTH=')F7.4x 1X, 'M')T30; 'HAND VOL.=';
     & F7.4,1X,'M CUSE')
       WRITE(3,34)HL,VL
34
       FORMAT (1X)*LEG LENGTH=*/67.4/1X/*M*/T39/*LES VOL.=*/
      & f7.4, 1x, 'M (URE!)
       WRITE(3,35)HF,VF
35
       FORMAT (1X, FOOT LENGTH= ',F7.4,1X, 'M',T3D, 'FOUT VOL.=',
     8 F7.4, 1x, 'M CUBE')
       WRITE(3,181)
       FORMAT (/, * FOLLOWING ARE S(I) VALUES IN SU M')
181
       WRITE(3,51)(S(1),1=1,6)
51
      FORMAT (1(6F8.4,/))
       WRITE(3,183)
       FORMAT (// FOLLOWING ARE AL (1) VALUES IN M')
183
      WRITE(3,19) (AL(1),1=1,6)
19
      FORMAT (1(6F8 -4,/))
      WRITE(3,185)
185
      FORMAT (// FOLLOWING ARE DO (I) VALUES IN "1")
       WRITE(3,20)(DO(1),1=1,6)
20
      FORMAT (1(6F8.40/))
```

```
WRITE(3,187)
      FORMAT(/, FOLLOWING ARE HSS(I) VALUES IN KCAL/HR/DEG C*)
187
      WRITE(3,18)(HSS(1),1=1,6)
18
      FORMAT (1(6F8.4//))
      WRITE(3,189)
189
      FORMAT(/, FOLLOWING ARE TC(I) VALUES IN KCAL/HR DEG C*)
      WRITE(3,17)(TC(1),I=1,60)
17
      FORMAT (6(10F7.2./))
      WRITE(3,191)
      FORMAT ( / , * FOLLOWING ARE A ( I ) VALUES IN SQ M *)
191
      WRITE(3,163) (A(I),I=1,60)
163
      FORMAT (6(1)F7.2,/))
      WRITE(3,193)
      FORMAT(/, FOLLOWING ARE RMP(I) VALIES IN M*)
193
      WRITE(3,164) (RMP(1),1=1,60)
164
      FORMAT (7(10F7.4//))
      WRITE(3,195)
195
      FORMAT(/, FOLLOWING ARE X(I) VALUES IN M')
      WRITE(3,165)(x(1),1=1,60)
165
      FORMAT (7(10F7.4//))
      WRITE(3,197)
197
      FORMAT(/, FOLLOWING ARE THE VALUES OF CS(1) AND C3(1)')
      WRITE(3,47)CS(1),CB(1)
      FORMAT (1x, 'CS(1) = ', F7.3, T30, 'CB(1) = ', F7.3)
47
      WRITE(3,199)
199
      FORMAT(/,' FOLLOWING ARE C(1) VALUES IN KCAL/DEG-C')
      WRITE(3,167)(C(1),1=1,66)
167
      FORMAT (7(10F7.3,/))
      WRITE(3,201)
201
      FORMAT(/, FOLLOWING ARE QB(I) VALUES IN KCAL/HR*)
      WRITE(3,169)(QB(I),I=1,60)
169
      FORMAT (7(10F7.3,/))
      WRITE(3,203)
203
      FORMAT (/, FOLLOWING ARE BFB (I) VALUES IN L/HR')
      WRITE(3,171)(RFP(I),I=1,60)
171
      FORMAT (7(10F7.3,/))
      WRITE(3,205)
      FORMAT(/, FOLLOWING ARE R(I) VALUES IN M 1)
205
      WRITE(3,173)(R(1),I=1,66)
173
      FORMAT (7 (10 F7.3 //))
      WRITE(3,207)
207
      FORMAT(/, FOLLOWING ARE CM(I) VALUES IN M')
      WRITE(3,175)(CM(I),I=1,66)
175
      FORMAT (7(10F7.4//))
      WRITE(3,209)
209
      FORMAT ( / , ' FOLLOWING ARE SWT (1) VALUES IN KG ')
      WRITE(3, 177) (SWT(1), 1=1,61)
177
      FORMAT (7(10 F7.3./))
      RETURN
      END
```

```
SUBROUTINE WETMAN
           REAL MU
            DIMENSION T(67), TSET(61), RATE(61), F(67)
           DIMENSION WARM(61), COLD(61), HF(67), ERROR(61)
           DIMENSION ER (60),Q(60),E(60),BF(60)
           DIMENSION TO (60) BC(60)
           DIMENSION P(11), SKINS (6), SKINV (6), SKINC (6)
           DIMENSION RF (6), HO (6), U (6)
           DIMENSION SKINR (6)
           DIMENSION WORKM(6), CHILM(6), H(6), PSKIN(6), EMAX(6)
           DIMENSION ATIME(9), ATAIR(9), AV(9), ARH(9), AWD RK(9)
           DIMENSION A(238), B(288)
            COMMON / PASS / AL (6) , DO (6) , HSS (6) , S(6) , TC (60) , Q &(6U) , R FP (60)
          *,C(67),CS(1),CB(1),RK(6)
           JATA ATIME/0.,30.,60.,90.,120.,150.,180.,210.,250./
           DATA ATAIR/1.7,1.7,1.7,1.7,1.7,1.7,1.7,1.7,1.7,1.7/
           C
           1. 121,.22 1,.22 1,.22 1,.22 1,.122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 .,122 
           DATA ARH/0.3.0.3.0.3.0.3.0.3.0.3.0.3.0.3.0.3/
           DATA AWORK/(1.,0.,0.,(1.,0.,0.,0.,0.,0.,0.,0.,
           DATA P/4.579,6.543,9.209,12.788,17.535,23.756,31.824,42.175,
          3 55.324,71.86,92.51/
           DATA TSFT/36.96,36.96,36.96,36.96,35.07,35.07,35.07,35.07,35.07
         3 34.81,34.58,36.89,36.89,36.89,36.89,36.28,36.28,36.28,36.28,36.28,
         8 34.53, 33.62,35.53,35.53,35.53,35.53,35.4.12,34.12,34.12,34.12,34.12,
         k 33.59, 33.25, 35.41, 35.41, 35.41, 35.41, 35.38, 35.38, 35.38, 35.38, 35.38,
         w 35.30,35.22,35.81,35.61,35.81,35.81,35.81,35.3,35.3,35.3,35.3,35.3,35.31,
         & 34.17,35.14,35.14,35.14,35.14,35.14,35.03,35.03,35.03,35.03,35.03,35.03,35.11,
         8 35.04,36.71/
           DATA EB/60*0./.RATE/61*0./
           DATA SKINR/9.0695/0.4935/0.0686/0.1845/0.1505/0.0334/
           DATA SKINS/U.081,0.481,0.154,0.031,0.218,0.35/
           DATA SKINV/0.132/0.322/0.095/0.121/0.230/0.10/
           DATA SKINC/0.05,0.15,0.05,0.35,0.05,0.35/
           DATA HORKM/C.O.O.O.075,0.02,0.0025,0.15,0.0025/
           DATA CHILM/0.005/0.2125/0.0125/0.0/0.0175/0.0/
           DATA CSW/320.0/,SSW/29.0/,PSW/0.0/
           DATA CDIL/117.0//SDIL/7.5//PDIL/0.0/
           DATA CCHIL/0.0//SCHIL/0.0//PCHIL/21.0/
           DATA CCON/5. 1/2 SCON/5.0/2PCON/0.0/
           DATA BULL/10.0/
           8 35.,35.,35.,35.,35.,35.,35.,35.,34.,34.,34.,34.,
         8 35.81,35.81,35.81,35.81,35.81,35.3,35.3,35.3,35.3,35.3,35.31,34.1,
         8 34.58,33.,33.,33.,34.,32./
           DATA 3M/39.4/
           KK =
           JJ=68
           DO 99 I=1.6
99
           RATE (10 * I) = 0.03
           READ (1,5) PR, MU, SV, COND
```

```
5
      FORMAT (4F13.5)
       READ(1,6) TAM, PRA, MUA, SVA, CONDA
       FORMAT (5F17.5)
6
       SA=0.3
       DO 110 K=1.6
110
       SAESA+S(K)
       TIME=0.0
       INT=5
       ITIME= "
       JTIME= D
       00 172 N=1.61
       F(N)=0.0
102
       CONTINUE
      1 = 1
199
       IF(ITIME.LT.ATIME(L+1)) GO TO 189
      L ≈ L + 1
      GO TO 199
189
      TAIR=ATAIR(L)
       V=AV(L)
       RH=ARH(L)
      WORK = 4 WORK (L)
       IF(WORK-(HM + SA)) 104,104,105
1/)4
      WORK I = 0 . 0
      60 TO 166
105
      WORKI = (WORK = (BM * SA)) * 0.78
106
      CONTINUE
      DENA=1.0/SVA
      RE(1)=((DENA*V*DO(1))/MUA)
      HO(1)=(CONDA/DO(1))*(0.97+0.68*(RE(1)**0.5))*(PRA**3.31)
      00 2 1=2.6
      DEN=1.0/SV
      RE(1)=((DEN * V * DO(1))/MU)
      HO(I)=(COND/DO(I))+(0.60+(RF(I)++0.50)+(PR++0.31))
2
      CONTINUE
      HO(1) = (COND/DO(1)) * (0.97+0.68 * (RE(1) * * 0.5)) * (PR* * 0.3)
      U(1)=1.0/((1.0/h0(1))+(AL(1)/RK(1)
      U(2)=1.0/((1.0/h0(2))+(AL(2)/RK(2)
      U(3)=1.0/((1.0/PO(3))+(AL(3)/RK(3))
      U(4)=1. 9/((1.0/HO(4))+(AL(4)/RK(4)
      U(5)=1.0/((1.0/PO(5))+(AL(5)/RK(5)
      U(6)=1.C/((1.0/HO(6))+(AL(6)/RK(6))
      DO 4 I=1.6
      H(I) = H(I) * S(I)
      CONTINUE
      I=(TAM/5)+1.0
      PIN=P(I)*(P(I+1)-P(I))*(TAM-5*(I-1))/5.0
      PAIR=RH*PIN
      K = (T(11)/5) + 1, 9
      POUT=P(K)+(P(K+3)-P(K))*(T(11)-5*(K-1))/5
3.01
      CONTINUE
      DO 302 N=1,61
      WARM(V)=0.0
      COLD(Y) =0.0
      IF (F(N)) 31%311,311
```

```
311
       f(N) = 0.0
310
       CONTINUE
       ERROR(N) = T(N) = TSET(N) + RATE(N) + F(N)
       IF (ERROR(N)) 303,302,304
323
       COLD(N) =- IRROR(N)
       60 TO 302
304
      WARH(N) = ERROR(N)
302
      CONTINUE
       WARMS=0.0
      COLDS=0.1
      DO 305 1=1.6
      K=10+1
      WARMS-WAPMS+WARM(K) + SKINR(I)
      COLDS=COLDS+COLD(K)*SYTYR(I)
305
      CONTINUE
      SWEAT= CSW*ERROR(1)+SSW*(VARMS-COLDS)+PSW*WARM(1)*WARMS
      DILAT= CDIL*ERPOR(1)+SDIL*(WARMS-COLDS)+PDIL*64RM(1)*WARMS
      STRIC= - CCON+EKROR(1) - SCON+(WARMS-COLDS)+FCON+FOLD(1)+COLDS
C THIS CONTROLLER PASED ON FULL SUIT DATA
      CHILL=BM+1.98703250-8.92(95166+(T(11)-TSET(11))
C
     $-2.6535 6692 * (T(60)-TSET(60))-4.59453410 * (T(50)-TSET(50))
     $-5.36126819 * (T(30) - TSET(30))
C THIS CONTROLLER BASED ON PRELIMINARY DATA
      CHILL= 7.001-5.822*(T(50))-TSE1(50))-2.407*(T(30)+TSET(30))
     #-37.338 * (T(11) - TSET(11))
      IF (SWEAT) 309,312,312
309
      SKEAT=0.0
      IF(DILAT) 313,314,314
312
313
      DILAT= 0.0
314
      IF (STRIC) 315,316,316
315
      STRIC: 0.0
316
      IF(CHIL) 317,318,314
317
      CHILL=0.0
318
      CONTINUE
4 .0
      CONTINUE
      DO 401 1=1.6
      N=10+1-9
      Q(N)=QB(N)
      BF(N) = BFB(N)
      (i). (=(i/) 3
      Q(N+1) = QB(N+1)
      BF(N+1) = BFB(N+1)
      F(N+1)=0.0
      Q(N+2) = QB(N+2)
      BF(N+2)=BFB(N+2)
      E(N+2) = 0.0
      Q(N+3) = QR(N+3)
      BF(N+3) = BFH(N+3)
      F (N+3) = 0.0
      Q(N+4) = QP(N+4) + WORKY(I) *WORK I + CHILM (I) *CHILL
      BF(N+4) = BFB(N+4) + Q(N+4) - QB(N+4)
      F (N+4) = 0_0
      Q(N+5) = QP(N+5) + PORKM(I) + WORK I+ CHILM(I) + CHILL
      BF(N+5)=BFB(8+5)+Q(N+5)-QB(Y+5)
```

```
E(N+5) = 0.0
       Q(N+6) = QB(N+6) + WORKM(I) + WORKI+ CHILM(I) + CHILL
       BF(N+6) = BFB(N+6) + Q(N+6) - QB(N+6)
       F(N+6) = 0.0
       Q(N+7) = QB(N+7) + WORKM(I) + WORK I + CHILM(I) + CHILL
       PF(N+7) = BFB(N+7) + R(N+7) - RB(N+7)
       E(N+7) = 0.0
       Q(N+B) = QB(N+B)
       BF(N+8)=BFB(N+8)
       E(N+8) = 0.0
       Q(N+9) = OP(N+9)
       &F(N+9)=((BFP(N+9)+SK1NV(I)*DILAT)/(1.0+SKIVC(1)*
      & STRIC)) +2.0 ++ (ERROR (N+9)/6.0)
       E(1+9)=(1.0
4 111
       CONTINUE
       IF(AWORK(L)-(SA*BM)) 14,14,15
14
       BWORK=SA*BM
       60 TO 16
15
       BWORK=AWORK (L)
16
       CONTINUE
       VE = (22 _ D*BWORK)/4.83
       VEMIN=VE/60
       BP=763.0
       T1=TAIR+273.0
       TE=T(11)+273.0
       AIRDCP = (0.2905 * (BP/760) * (293/TE))/1000
      H2 DOUT = VE*((PP*99.962)/(62.4*TE))*(POUT/(BP-POUT))*
     & (18.01534/99.962)
       RHEAT= (6.7E-3*(VEMIN**2))+(6.2F-5*(VEMIN**3))
       RHL=(VE*AIRDCP*(TE-TI))+(0.58*H200UT)
       DO 602 1=11,14
       F(I) = (RHL - 0.06 * RHEAT) / 4.0
6 95
      CONTINUE
      00 500 K=1,60
      BC(K)=BF(K)*(T(K)-T(61))
      TD(K) = TC(K) * (T(K) - T(K+1))
500
      CONTINUE
      00 5(1 1=1.6
      K=10*I-9
      HF(K)=Q(K)-E(K)-BC(K)-TD(K)
      HF(K+1) = O(K+1) - E(K+1) - BC(K+1) - TD(K+1) + TD(K)
      HF(K+2) = Q(K+2) - F(K+2) - BC(K+2) - TD(K+2) + TD(K+1)
      HF(K+3) = Q(K+3) - E(K+3) - BC(K+3) - TD(K+3) + TD(K+2)
      HF(K+4) = Q(K+4) + F(K+4) + BC(K+4) + TD(K+4) + TD(K+3)
      HF(K+5)=0(K+5)-E(K+5)-BC(K+5)-TD(K+5)+TD(K+4)
      HF(K+6)=G(K+6)-F(K+6)-PC(K+6)-TP(K+6)+TP(K+5)
      HF(K+7)=Q(K+7)-E(K+7)-BC(K+7)-TD(K+7)+TD(K+6)
      HF(K+B)=Q(K+8)-E(K+8)-EC(K+8)-TD(K+8)+TD(K+7)
      HF(K+9) = O(K+9) - E(K+3) - BC(K+9) + TD(K+8) - HSS(I) * (T(K+9) - T(I+61))
      HF(I+61)=HSS(!)*(T(K+9)-T(I+61))-H(I)*(T(I+61)-TAIP)
571
      CONTINUE
      HF(61)=0.0
      DO 572 K=1,60
      PF(61) = HF(61) + PC(K)
502
      CONTINUE
```

```
DT=0.01666666666667
          00 670 K=1.67
          F(K)=4F(K)/C(K)
         AU=ABS (F(K))
         1F(AU+DT-0.1) 600,607,601
   6.1
         PT=0.1/AU
         CONTINUE
  600
         00 7 3 K=1.67
         T(K)=T(K)+F(K)*DT,
  700
         CONTINUE
         TIME = TIME + DT
         LTIME=60. *TIME
         IF(LTIME-INT-ITIME) 301,701,701
  701
         CONTIVUE
         ITIME=ITIME+INT
         0.3 = 0.3
        C. 0=9H
        E V=0.0
        TS=0.3
        T5=0.0
        HfLOW= (). c
        $8F=0.0
        28.1=H 0'3 00
        CC=CO+BF(N) /60.0
        HP=HP+Q (N)
        FA=EA+E(M)
 8 00
        CONTINUE
        00 892 1=1.6
        SPF=SRF+PF(10*1)/60.0
        TS=TS+T(10+1)+C(10+1)/CS(1)
 8^2
        CONTINUE
       DG 8 11 N=1,61
       TB=TB+T (H) + C (N) / C9 (1)
       HFLOW=HFLOW+HF(N)
 8:1
       CONTINUE
       EV=EV/SA
       HF=HP/SA
       HFLOW=HFLOW/SA
       FF=E(11)+E(17)+E(13)+E(14)
       CONDZ= (HP-FE/SA-HFLOW)/(T(61)-TS)
       IF(ITIME-INT) 910,910,911
910
       WRITE(7,999)
999
       FCRMAT (*1", 1CX, BAYLEY PVC C.R.")
       WRITE(3,912)
       FORMAT ( * '-1x, 'T', 4x, 'TW', 2x, 'HFLW', 2x, 'M', 5x, 'FV', 3x, 'TE', 4x, 'TS'
912
     8,4x, "TH ",4x, "TO ',4x, "TP ",4x, "TM ",3x, "SBF ",3x, "CO ",2x, "CCND ",2x,
     & TLEG', ZX, TARM', ZX, TTRNK')
      NN= :
911
      WRITE(3,915) IT INE, TAIR, HELOW, HP, EV, TB, TS, T(1), T(61), T(11),
     8 T(47), SBF, CO, COND2, T(50), T(30), T(20)
      FORMAT ( 1 1, 13, F5. 1, 316. 1, 6F6. 2, F5. 2, 2F5. 1, 3F6. 2)
915
      50 2 156 1=1,6
```

```
COND3 = COND3 + H(I) + (I(I+6I) - IAIR)
2050
      CONTINUE
      COND 3= COND 3 / SA
      HIRAL=HP-HFLOW-FE/SA-COND3
      N 4 = NN + 1
C1130 KK=KK+1
C
      A(KK)=ITIME
C
      A(JJ+KK)=TAIR
C
      A(2*JJ+KK)=T(11)
C
      \Lambda(?*JJ+KK)=TS
C
      A(4*JJ+KK)=T(1)
C
      B(KK)=ITIME
      B(JJ+KK)=TAIR
C
      分(5*JJ+kK)=Hむ
C
      R(3*JJ+KK)=SEF
C
      B (4+JJ+K+)= C 0
      JTIME=JTIME + INT
      TECUTIME-30) 301,1101,1101
1101
      JTIME=U
      IF (ITIME-90) 102,1102,1102
C1132 CALL PLOTEE (1, A, 48, 5, 48, (1)
      WRITE (3,3001)
C3001 FORMAT ("-",1"X,"1=TUTR",5X,"2=TR",5X,"3=TS",5X,"4=TH")
      CALL PLOTEE (2,8,48,5,48,7)
      WRITE (3,3002)
C3002 FORMAT(!-!,10x,!1=TVTR!,5x,!2=M!,5x,!3:SBF!,5x,!4=C0!)
1102
      PETURY
      END
```

```
183.4900069300.0000

39.49600

9.55000 4.70800 0.09100 9.48400

21.00000 0.71000 0.96580 9.81620 0.35120
```

HODY S.G.=1.08462 % BODY FAT= 0.0711 SURFACE AREA= 1.881350 M

ADIPOSE TISSUE WT. = 4.9297 KG NON-ADIPOSE TISSUE WT. =64.3700 KG

HEAD RADIUS= 0.0992 M
TRUNK LENGTH= 0.9709 M
ARM LENGTH= 0.7635 M
HAND LENGTH= 1.1823 M
LEG LENGTH= 1.4424 M
FOOT LENGTH= 1.5493 M

HEAD VOL.= 0.0039 M CUHE TRUNK VOL.= 0.0377 M CUBE ARM VOL.= 0.0066 M CUBE HAND VOL.= 0.0006 M CUBE LEG VOL.= 0.0157 M CUBE FOOT VOL.= 0.0009 M CUBE

FOLLOWING ARE S(I) VALUES IN SQ M 0.1398 0.7161 0.2825 0.1409 0.6543 0.1905

FOLLOWING ARE AL(I) VALUES IN M 0.0031 0.0031 0.0029

FOLLOWING ARE DO(I) VALUES IN M
-0.2110 0.2349 0.1179 0.0380 0.1445 0.0392

FOLLOHING ARE HSS(I) VALUES 19 KCAL/HR/DFG C 1.5438 8.9372 3.3309 1.2735 7.8793 1.2441

FOLLOWING ARE TO(I) VALUES IN KCAL/HR DEG C 0.61 8.36 0.0 1.58 2.92 6.99 30.33 31.63 32.93 13.73 20.48 22.29 27.61 $C \cdot C$ 3.63 7.56 11.60 14.14 16.57 24.38 6.03 14.25 17.32 19.08 18.23 0.02.56 9.12 10.05 11.17 14.12 28.13 71.10 75.84 86.59 20.15 7.19 0.0 4.4? 9.33 0.0 5.40 11.39 17.23 19.12 21.43 27.24 33.74 40.11 41.88 5.80 12.23 18.51 40.14 150.35 156.56 162.78 27.71 10.48 0.0FOLLOWING ARE A(I) VALUES IN SQ M 0.08 0.04 0.06 0.10 0.11 0.11 0.11 0.11 0.12 0.13 0.21 0.37 0.59 0.37 7.43 0.49 0.54 0.66 0.63

0.68 0.13 0.220.26 0.07 0.10 0.15 0.17 0.200.23 0.24 0.03 0.07 2.34 0.05 0.76 0.05 0.07 0.07 0.08 0.17 0.17 0.25 0.31 0.52 0.58 0.47 0.56 3.36 0.42 0.61 0.08 9.99 0.10 0.10 0.14 9,04 0.06 7.09 7.10 0.11

FOLLOWING ARE RMP(I) VALIES IN M
0.0553 0.9715 C.0822 0.0891 0.0920 0.0931 0.0942 0.0954 0.0971 0.1004
0.9338 0.0491 0.0604 0.0703 0.0798 0.0887 0.0968 0.1037 0.1084 0.1122
0.0149 0.0217 0.0267 0.0313 0.0361 0.0408 0.0450 0.0464 0.0507 0.0537
0.0041 0.0059 0.0073 0.0083 0.0087 0.0090 0.0093 0.0097 0.0108 0.0139
0.0191 0.0277 0.0340 0.0399 0.0460 0.0519 0.0571 0.0613 0.0639 0.0670
0.0046 0.0067 0.0082 0.0092 0.0097 0.0099 0.0101 0.0105 0.0115 0.0145

```
FOLLOWING ARE X(I) VALUES IN M
3.0202 0.0122 0.0093 0.0046 0.0011 0.0011 0.0011 0.0014 0.0021 0.0045
0.0181 0.0125 0.0161 0.0097 0.0094 0.0084 0.0077 0.0061 0.0034 0.0042
0.0080 0.0055 0.0045 0.0048 0.0050 0.0044 0.0040 0.0028 0.0019 0.0042
0.0022 0.0015 0.0012 0.0007 0.0003 0.0003 0.0003 0.0006 0.0016 0.0045
0.0102 0.0370 0.0057 0.0060 0.0062 0.0055 0.0050 0.0033 0.0020 0.0043
FOLLOWING ARE THE VALUES OF CS(1) AND CB(1)
CS(1) = 3.438
                              CB(1) = 56.347
FOLLOWING ARE C(I) VALUES IN KCAL/DEG-C
 0.564
        0.554
                0.564
                       0.564
                               0.085
                                      0.085
                                             0.085
                                                     0.085
                                                            0.099
                                                                   0.245
 2.526
        2.506
                2.506
                       2.506
                               4.105
                                      4.105
                                             4.105
                                                     4.105
                                                            1.873
                                                                   1.234
                                      0.772
 0.362
        0.362
                0.362
                       0.362
                               0.772
                                             0.772
                                                     r.772
                                                            0.256
                                                                   0.443
 0.736
        0.036
                0.036
                       0.936
                               C.017
                                             0.017
                                      0.017
                                                     0.017
                                                            0.039
                                                                   0.170
        1.077
                                                            1.631
 1.077
                1.077
                       1.377
                               2.331
                                      2.331
                                             2.331
                                                     2.331
                                                                   1.097
 0-761
        0.051
                0.061
                       0.061
                               0.017
                                      0.017
                                             0.217
                                                     0.017
                                                            0.059
                                                                   0.218
 2.250
        0.050
                0.265
                       0.102
                               0:045
                                      0.238
FOLLOWING ARE QB(I) VALUES IN KCAL/HP
 3.167 3.167
               3.167
                      3.167
                              0.031
                                      0.031
                                             0.031
                                                     0.031
                                                            0.049
                                                                   0.782
11.364 11.364 11.364 11.364
                               1.502
                                      1.502
                                             1.502
                                                     1.502
                                                            0.937
                                                                   0.411
 0.151
        0 - 15 1
                0.151
                       0.151
                              9.283
                                      0.283
                                             0.283
                                                     0.283
                                                            0.128
                                                                   0.148
        0.018
 0.318
                0.018
                       3.918
                              0.006
                                      0.006
                                                                   0.057
                                             0.006
                                                    0.006
                                                            0.050
        0.466
                0.466
 2.466
                       9.466
                               7.853
                                      0.853
                                             0.853
                                                     0.853
                                                            0.315
                                                                   0.366
 0.029
        0.029
                0.029
                       0.029
                              0.006
                                      0.006
                                             0.006
                                                     0.006
                                                            0.030
                                                                   0.073
FOLLOWING ARE BES (I) VALUES IN L/HR
11.250 11.250 11.250 11.250
                                                    0.037
                              0.037
                                      0.037
                                             0.037
                                                            1.059
                                                                   1.455
                              1.893
52.500 52.500 52.500 52.510
                                             1.803
                                      1.803
                                                    1.863
                                                            1.124
                                                                   2.138
       0.181
 3.181
                0.181
                       0.131
                              0.339
                                      0.339
                                             0.339
                                                    5.339
                                                            9.154
                                                                   0.665
 9.021
        0.021
                9.021
                       0.021
                                                            0.024
                              0.007
                                      0.007
                                             0.707
                                                    (.007
                                                                   1.056
 0.560
        0.560
                0.560
                       0.560
                              1.324
                                      1.024
                                             1.024
                                                    1.024
                                                            0.379
                                                                   1.024
 0.035
        0.035
                0.035
                       0.035
                              0.007
                                      C.007
                                             0.007
                                                    0.007
                                                            0.035
                                                                   0.567
FOLLOWING ARE R(I) VALUES IN M
        9.072
 0.057
               0.082
                       0.091
                              0.092
                                      0.093
                                             0.794
                                                    0.395
                                                            0.097
                                                                   0.099
        5.049
 0.335
                0.061
                       0.070
                              0.080
                                      930.0
                                             0.097
                                                    0.104
                                                            6.100
                                                                   0.111
        0.022
                9.927
 0.015
                       0.031
                              7.035
                                      0.041
                                             0.045
                                                    0.049
                                                            0.051
                                                                   0.053
        0.006
               0.907
                       2.078
                                             0.009
 0.054
                              0.009
                                      0.009
                                                    0.010
                                                            0.010
                                                                   7.013
                                             0.757
 0.020
        0.028
               0.034
                       0.039
                              5.045
                                     0.052
                                                    0.062
                                                            0.064
                                                                   0.066
 0.005
        0.007
                0.008
                       0.010
                              0.010
                                     0.010
                                             0.010
                                                    0.010
                                                            0.011
                                                                   0.013
               0.059
                       0.019
                                     0.020
 0.105
        9.117
                              0.072
FOLLOWING ARE CM(I) VALUES IN M
0.0452 0.0654 0.0776 0.0868 0.0914 0.0925 0.0936 0.0947 0.0961 0.0982
0.0247 0.0428 0.0553 0.0654 0.0751 0.0845 0.0929 0.1006 0.1067 0.1101
0.0109 0.0189 0.0244 0.0289 0.0337 0.0386 0.0430 0.0470 0.0497 0.0516
0.0030 0.0052 0.0067 0.0079 0.0086 0.0089 0.0092 0.0094 0.0100 0.01%
7.9139 2.3242 0.3312 9.0369 0.0429 0.0491 0.0546 0.0597 9.0629 9.0649
0.0034 0.0058 0.0075 0.0089 0.0096 0.0098 0.0100 0.0102 0.0108 0.018%
0.1026 0.1143 0.0559 0.0161 0.0691 0.0167
```

	COND ILES TARM TTRNK -9 37.4 31.12 36.55 31.39 -3 21.4 28.89 28.41 28.87 -6 17.6 27.57 27.30 27.52 -8 15.6 26.24 26.12 26.23 -4 13.0 25.40 25.54 25.71 -5 12.0 25.40 25.54 25.71 -5 12.0 25.40 25.57 25.71 -5 12.0 25.40 25.57 25.71 -6 17.3 24.36 25.27 25.40 -9 11.8 24.36 25.27 25.37 -9 11.8 24.36 25.03 25.15 -1 11.7 24.12 25.03 25.15 -1 11.7 24.12 25.03 25.15 -1 11.7 24.12 25.03 25.15 -1 11.5 23.86 24.99 25.01 -1 11.5 23.86 24.99 25.01
0.164 0.273 3.122 1.371 0.427 0.492 0.066 0.189 1.057 1.219 0.009 0.242	35.15 0.08 4 35.56 0.02 5 33.88 0.01 5 32.88 0.01 6 31.41 0.01 6 31.67 0.01 6 30.49 0.01 7 30.49 0.01 7 30.25 0.01 7
95 0.095 61 4.561 58 0.858 19 0.019 90 2.590 19 0.019	6.93 37.20 6.49 35.76 6.28 35.76 6.28 35.76 6.28 35.76 5.42 35.34 5.42 35.81 5.42 35.83 4.53 35.81 4.62 35.83 4.63 35.83 4.63 35.83 4.63 35.83 4.63 35.83 4.63 35.83 4.63 35.83 4.63 35.83
0.095 4.561 6.858 0.858 0.019 0.03 0.019 0.019	3. 3. 3. 3. 3. 3. 3. 3. 3. 3. 3. 3. 3. 3
ALUES IN KG 0.765	6.8 6.8 6.8 6.8 6.9 6.9 6.9 6.9 6.9 6.9 6.9 6.9
MG ARE SWT(I) V G.765 3.730 3.730 3.730 0.572 0.572 0.572 0.572 0.572 0.572 0.110 0.110 0.110	BAYLEY PVC H HFLW M 7-145.5 42.8 7-119.1 55.9 7-101.7 64.4 7-88.5 71.3 7-78.1 77.0 7-62.1 86.3 7-55.0 90.1 7-55.0 90.1 7-55.0 90.1 7-55.0 90.1 7-55.0 90.1 7-55.0 109.0 7-24.0 110.9 7-25.4 110.9
FOLLOWING	- 24 - 25 - 25 - 25 - 25 - 25 - 25 - 25 - 25

To Run the Model

- Specific gravity of the simulate subject may be entered by data statement. If SG=0 the program will calculate SG based on subject's height and weight.
- 2. Protective device thickness (THWS) In meters is entered by data statement.
- Protective device thermal conductivity for each body segment (RK(N))
 is entered by data statement.
- 4. Water temperature (ATAIR) is entered by data statement.
- 5. Subject height and weight are read from the first data card.
- 6. The output may be labeled by simulated experiment by altering the hollerith stream in format statement 999.

[#] U.S.G.P.O. 620-939/1202-774